

## CCC Annual Report

UIUC, August 14, 2013

# *Modeling and Simulation of Multiphase Flows in CC Mold Region*

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## Outline

- **Determination of slide-gate position (PART 1)**
    - Using a gate-position-based flow rate model to back-calculate gate position based on measured casting speed and mold dimensions;
  - **Gas flow through UTN and bubble size (PART 2)**
    - To predict hot argon flow rate entering steel stream;
    - To estimate active sites number density at UTN inner surface
    - To predict initial bubble size
  - **Multi-phase flow modeling (PART 3)**
    - Board 4
    - Board 11~13
    - Board 14~16
- and Validation with nailboard measurements:**
- Surface velocity profile
  - Surface level profile
- also, Parametric study on bubble size effects**

# Acknowledgements

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- Continuous Casting Consortium Members (ABB, ArcelorMittal, Baosteel, Magnesita Refractories, Nippon Steel, Nucor Steel, Postech/ Posco, Severstal, SSAB, Tata Steel, ANSYS/ Fluent)
- **J. Powers and T. Henry in Severstal for the help with the plant nail board trials on Oct. 15~16 in Severstal in Dearborn, MI, the caster of which has:**
  - a 203 mm mold thickness
  - plant operation data recorded for gate position, flow rate, mold level, casting speed, ...
- **R. Singh for the help with nail board experiments**
- **Mihir Chavan for the nail board measurements**

## PART 1: Gate Position Estimation

# Modeling SEN Flow Rate

--Analysis of Bernoulli's Equation

$$\frac{p_0}{\rho g} + \frac{v_0^2}{2g} + z_0 = \frac{p_3}{\rho g} + \frac{v_3^2}{2g} + z_3 + h_{port} + h_f + h_{sg}$$

$$\frac{p_0 - p_3}{\rho g} + z_0 - z_3 = \frac{v_3^2}{2g} + \sum h \rightarrow h_{port} + h_f + h_{sg}$$

$$-H_3 \quad H_1 + H_2 + H_3 \quad \frac{v_{SEN}^2}{2g} \left( \frac{A_{SEN}}{A_{port}} - 1 \right)^2 \quad f \frac{L_{SEN}}{D_{SEN}} \frac{v_{SEN}^2}{2g}$$

$$h_{sg} = \left( \left( \frac{1}{\mu} - 1 \right)^2 \left( \frac{A_{SEN}}{A_{GAP}} \right)^2 + \left( \frac{A_{SG}}{A_{GAP}} - \frac{A_{GAP}}{A_{SG}} \right)^2 \left( \frac{A_{SEN}}{A_{SG}} \right)^2 \right) \frac{v_{SEN}^2}{2g}$$

$$\mu = \frac{A_{vc}}{A_{SG}} = \begin{cases} 0.63 + 0.37 \left( \frac{A_{GAP}}{A_{SG}} \right)^3 \\ 0.5864 + 0.2762 \left( \frac{A_{GAP}}{A_{SG}} \right) - 0.4807 \left( \frac{A_{GAP}}{A_{SG}} \right)^2 + 0.618 \left( \frac{A_{GAP}}{A_{SG}} \right)^3 \\ 0.64 \end{cases}$$

correlation 1<sup>[1]</sup>

cor. 2<sup>[2]</sup>

cor. 3<sup>[3]</sup>

Ref:

[1] Oertel, Herbert; Prandtl, Ludwig, et.al, Prandtl's Essentials of Fluid Mechanics, Springer, ISBN 0387404376. See pp. 163–165.

[2] Evangelista Torricelli, 1643

[3] [http://en.wikipedia.org/wiki/Vena\\_contracta](http://en.wikipedia.org/wiki/Vena_contracta)

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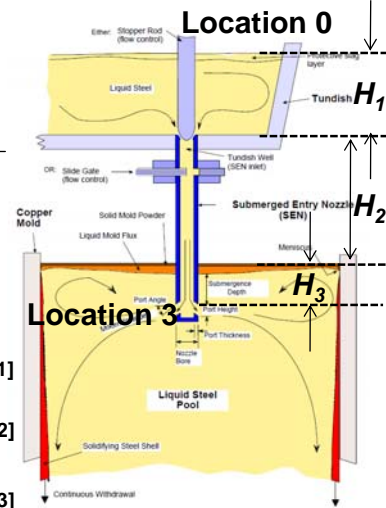
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$$p_0 = 1.01 \times 10^5 \text{ Pa}$$

$$p_3 = \rho g H_3 + p_0$$

$$z_0 - z_3 = H_1 + H_2 + H_3$$



## Gate-position-based Model (considering gas addition)

- Equation for gate-position-based model <sup>[1]</sup>:

$$Q_{SEN} = A_{eff} \sqrt{\frac{2g(H_1 + H_2)}{\left( \frac{A_{SEN}}{A_{port}} - 1 \right)^2 + f \frac{L_{SEN}}{D_{SEN}} + \left( \frac{1}{\mu} - 1 \right)^2 \left( \frac{A_{SEN}}{A_{GAP}} \right)^2 + \left( \frac{A_{SG}}{A_{GAP}} - \frac{A_{GAP}}{A_{SG}} \right)^2 \left( \frac{A_{SEN}}{A_{SG}} \right)^2 + \left( \frac{A_{SEN}}{2A_{port}} \right)^2}}$$

where  $\mu = 0.63 + 0.37 \left( \frac{A_{GAP}}{A_{SG}} \right)^3$

$$A_{eff} = \begin{cases} A_{SEN} & \text{single phase flow} \\ \frac{V_c WT}{Q_{gas} + V_c WT} A_{SEN} & \text{two phase flow} \end{cases}$$

For continuous caster, an extra term should be added to account for pressure drop due to clogging:

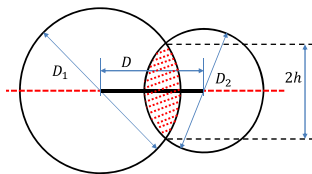
$$Q_{SEN} = A_{eff} \sqrt{\frac{2g(H_1 + H_2)}{\left( \frac{A_{SEN}}{A_{port}} - 1 \right)^2 + f \frac{L_{SEN}}{D_{SEN}} + \left( \frac{1}{\mu} - 1 \right)^2 \left( \frac{A_{SEN}}{A_{GAP}} \right)^2 + \left( \frac{A_{SG}}{A_{GAP}} - \frac{A_{GAP}}{A_{SG}} \right)^2 \left( \frac{A_{SEN}}{A_{SG}} \right)^2 + \left( \frac{A_{SEN}}{2A_{port}} \right)^2 + C}}$$

In current study,  $C=0$  is assumed (no clogging).

# Validation for Gate-position-based Model

--using full-scale water model measurements at ArcelorMittal, East Chicago

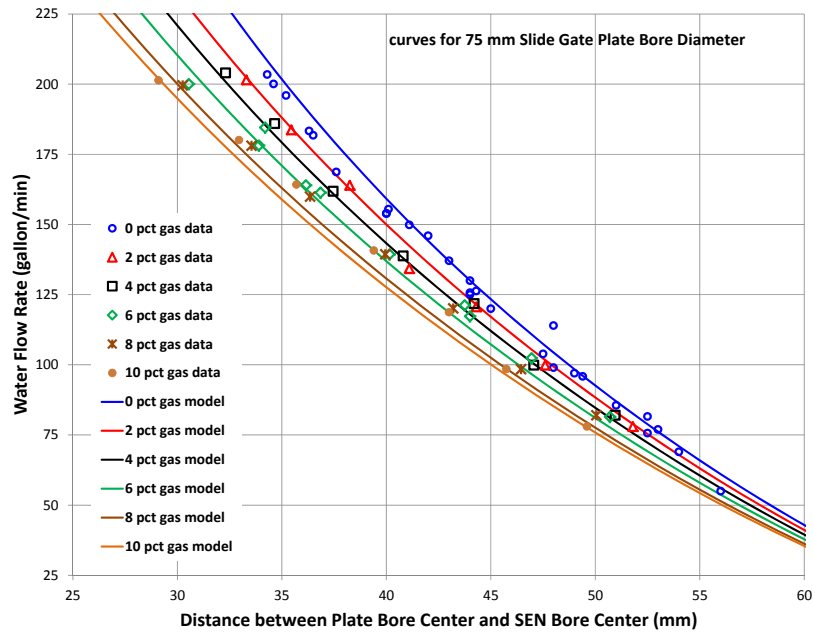
- Nice match is observed, analytical SEN flow rate model is validated
- Gap area is [1]:



$$h = \frac{D_1 D_2}{4D} \sqrt{1 - \left( \frac{D_1^2 + D_2^2 - 4D^2}{2D_1 D_2} \right)^2}$$

$$A_{GAP} = \frac{D_1^2}{4} \arcsin\left(\frac{2h}{D_1}\right) + \frac{D_2^2}{4} \arcsin\left(\frac{2h}{D_2}\right) - Dh,$$

$$\text{if } D > \frac{\sqrt{D_1^2 - D_2^2}}{2}$$



R. Liu, B.G. Thomas, B. Forman and H. Yin., AISTech Iron Steel Technol. Conf. Proc., 2012, Atlanta, GA, pp 1317.

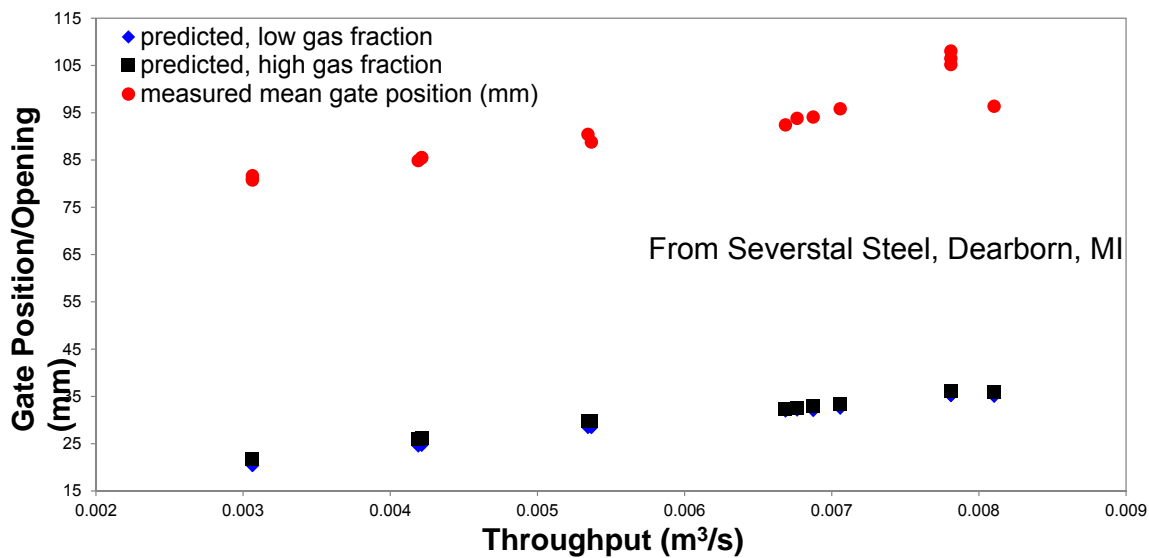
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## Predicted vs. Measured Gate Position at Severstal



Low/High gas fraction indicates the location where the fraction is calculated:  
Low fraction is calculated at UTN gas injection point;  
High fraction is calculated at SEN port exit.

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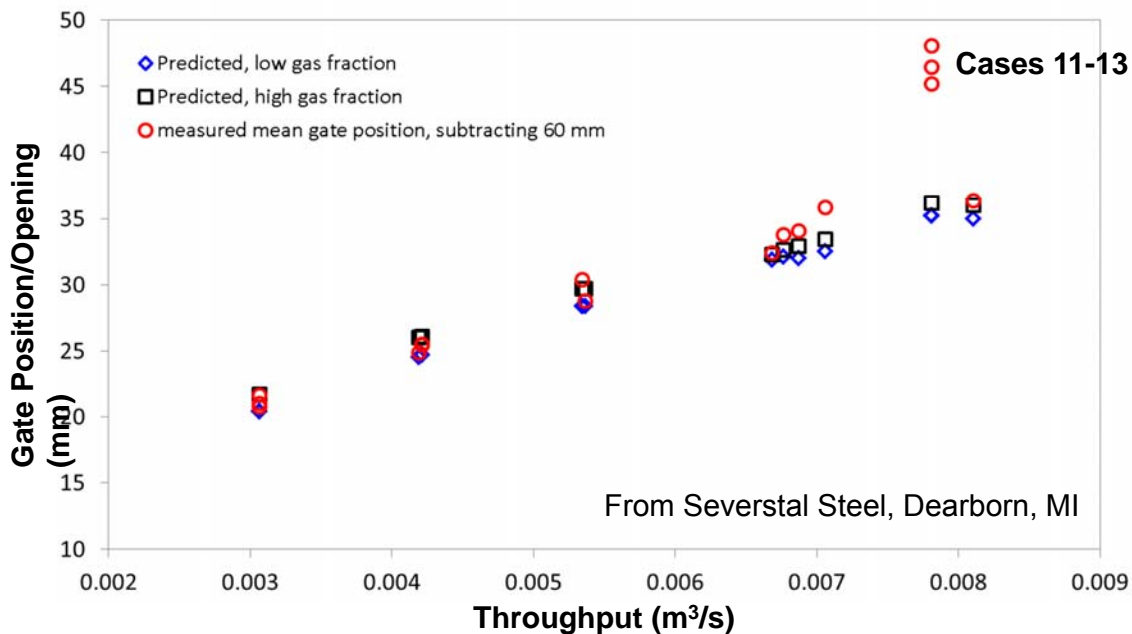
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# Predicted vs. Calibrated Measurements



Exact offset for quantitative gate position not recorded, but typically is 40-60mm

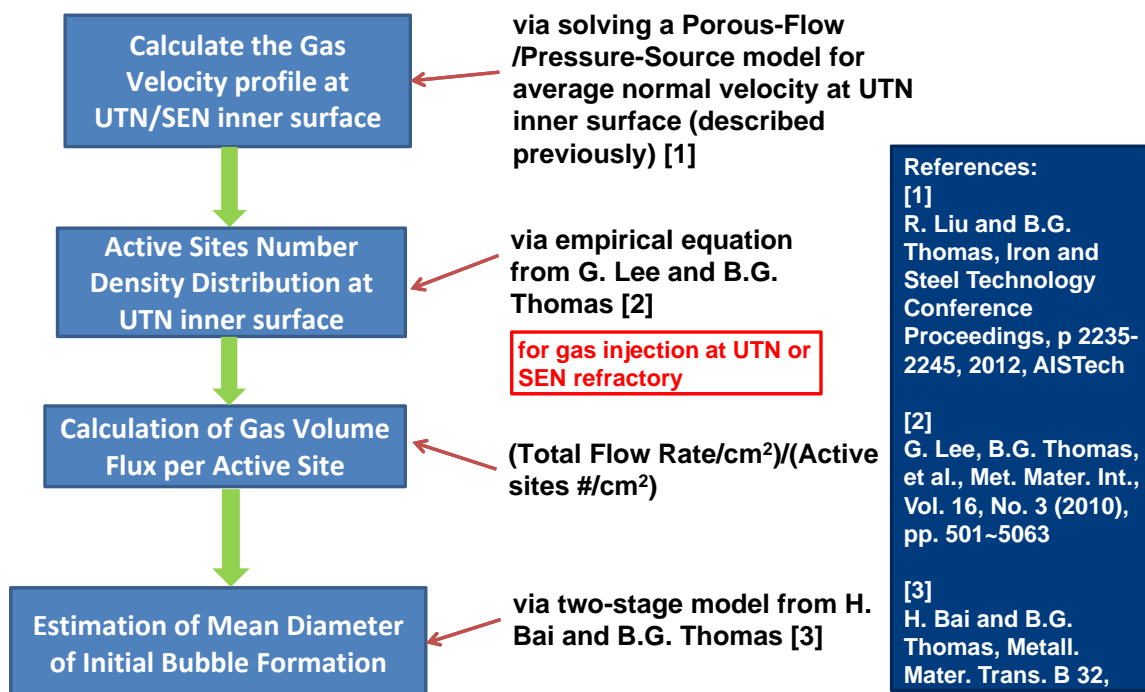
## Part 1. Gate-position Conclusions

- Prediction and measurements show the same trend at different throughputs;
- Once the measured gate position is calibrated by subtracting 60 mm, the predicted gate position usually matches closely with measurements; mismatch is restricted to one set of casting conditions (case 11~13);
- Gas flow (< 8%) has little effect on gate position (for range of conditions studied)

## PART 2:

# UTN Gas Flow Simulation and Initial Bubble Size Distribution

## Methodology of Initial Bubble Size Estimation in Argon-Steel System



# Governing Equations

## -- Pressure-Source Model

Darcy's Law:  $\mathbf{v} = -K_D \nabla p$  (1)

Mass Conservation (or continuity):  $\nabla \cdot (\rho \mathbf{v}) = 0$  (2)

Heat Conduction:  $\nabla \cdot (k \nabla T) = 0$  (3)

Ideal Gas Law:  $p = \rho RT$  (4)

Eqn (1) and (2)  $\Rightarrow \nabla \cdot (\rho K_D \nabla p) = 0 \Rightarrow \nabla \rho \cdot (K_D \nabla p) + \rho \nabla \cdot (K_D \nabla p) = 0$

$$\nabla \cdot (K_D \nabla p) = -\frac{1}{\rho} [\nabla \rho \cdot (K_D \nabla p)]$$

$$\rho = \frac{p}{RT}$$

$$\nabla \cdot (K_D \nabla p) = -\frac{RT}{p} \left[ \nabla \left( \frac{p}{RT} \right) \cdot (K_D \nabla p) \right] = \dot{S}$$

pressure diffusion      Pressure source due to gas expansion

### -- For Pressure-Source Model

- The **left hand side** of the final equation above (in red box) is the standard diffusion part, while the **right hand side** is in the form of a **source term** taking into account the effect of **gas expansion due to temperature and pressure gradient**.
- Using a 3-D FLUENT model, adding a source term to pressure (or a user-defined-scalar) diffusion equation coupling the energy equation

# Governing Equations

## -- Porous Flow Model

Mass Conservation (or continuity):  $\nabla \cdot (\rho \mathbf{v}) = 0$  (1)

Heat Conduction:  $\nabla \cdot (k \nabla T) = 0$  (2)

Ideal Gas Law:  $p = \rho RT$  (3)

Full Set of Navier-Stokes Equations for flow in porous media:

$$\nabla \cdot (\rho \mathbf{v} \mathbf{v}) \sim \frac{\rho V^2}{\Delta r} = \frac{0.3 \times 0.0073^2}{0.035} \approx 4.6 \times 10^{-4}$$

$$\frac{\partial (\rho \mathbf{v})}{\partial t} + \nabla \cdot (\rho \mathbf{v} \mathbf{v}) = -\nabla p + \nabla \cdot (\mu \nabla \mathbf{v}) + \mathbf{F} - \left( \frac{\mu}{\alpha} \mathbf{v} + C \frac{1}{2} \rho |\mathbf{v}| \mathbf{v} \right)$$

$$\nabla \cdot (\mu \nabla \mathbf{v}) \sim \frac{\mu V}{\Delta r^2} = \frac{8.1 \times 10^{-5} \times 0.0073}{0.035^2} = 4.8 \times 10^{-4}$$

(steady state problem)

viscous resistance      Re ~2.3, C=0 for laminar flow

inertial resistance

$$\frac{\mu}{\alpha} = \frac{1}{K_D}$$

$$\mathbf{v} = -K_D \nabla p$$

### -- Porous-Flow Model

- Solve a complete set of Navier-Stokes equations, with the momentum equation simplified into equation (1), adopting the ideal gas law to relate density with pressure and temperature.

# Permeability Formulation

- About permeability in Darcy's law:

- Permeability measured the viscous resistance of the porous media to fluid;
- Actual permeability consists of specific permeability and gas dynamic viscosity: specific permeability keeps constant for a chosen refractory and argon dynamic viscosity changes with local temperature.

$$K_D = \frac{K_{DS}}{\mu(T)} \quad \left\{ \begin{array}{l} K_{DS} = 1.01 \times 10^{-12} m^2 \quad (\text{constant specific permeability, from draft paper}) \\ \mu(T) \quad (\text{gas dynamic viscosity, as a function of local temperature}) \end{array} \right.$$

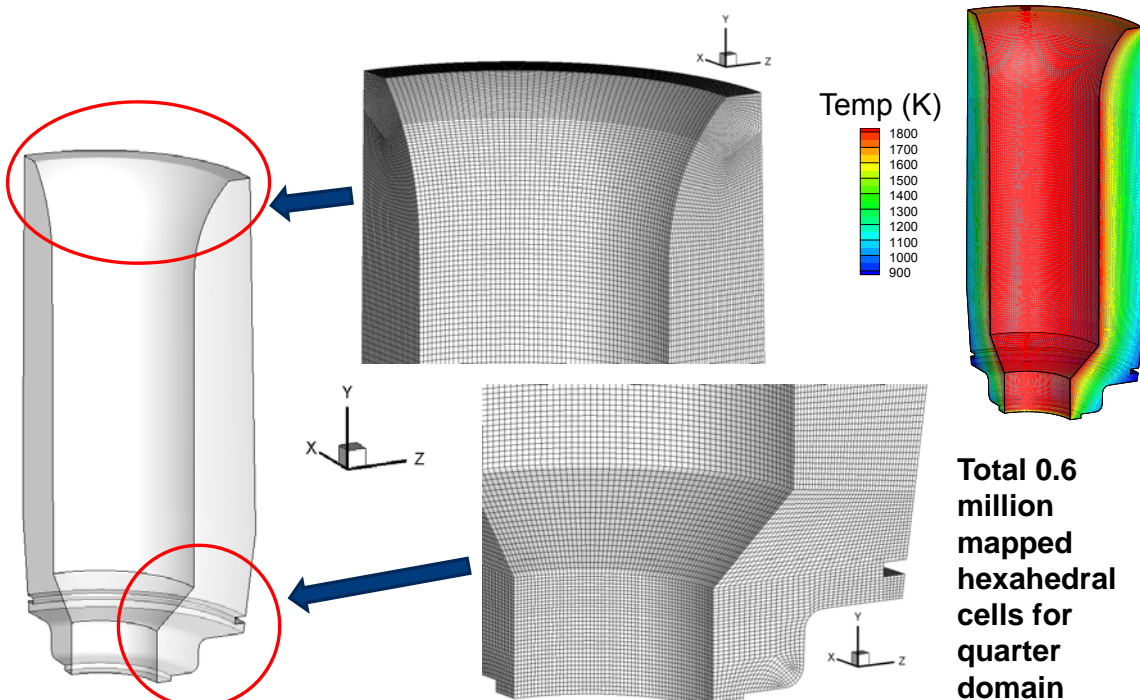
**Viscosity Varying with Temperature:**  $\log [\eta(T)/\eta(293)] = 0.63842 \log T - 6.9365/T - 3374.72/T^2 - 1.51196$  **Ref[1]**

**or**  $\mu(T) = \mu_0 * 10^{(0.63842 \lg T - 6.9365/T - 3374.72/T^2 - 1.51196)}$   $\mu_0 = 2.228 \times 10^{-5} Pa \cdot s$   
 Room temperature (20 C) argon viscosity

Ref:

[1] R. Dawe and E. Smith. Viscosity of Argon at High Temperatures. Science, Vol. 163, pp 675~676, 1969.

## Domain, Mesh and Temperature for Porous Gas Flow Model in Severstal UTN



# Parameters

Specific Permeability (npm)	Back Pressure (Psi)	Measured Argon Flow Rate (SLPM)
10.1	15.45	5.02

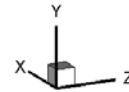
Boundary Conditions:

Parameters	Values
$h_{inner}$ (W/m <sup>2</sup> K)	$2.5 \times 10^4$
$h_{outer}$ (W/m <sup>2</sup> K)	10
Heat conductivity $K_s$ (W/mK)	33
$\mu$ (Pa*s)	0.0056
$\rho$ (kg/m <sup>3</sup> )	7200
Average steel velocity $U$ (m/s)	1.6

Symmetric B.C.

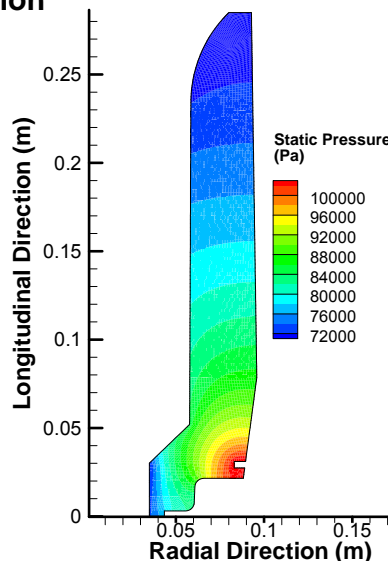
Pressure B.C. at UTN inner surface (with/without pressure correction using Bernoulli's equation)

Wall B.C. at anywhere else

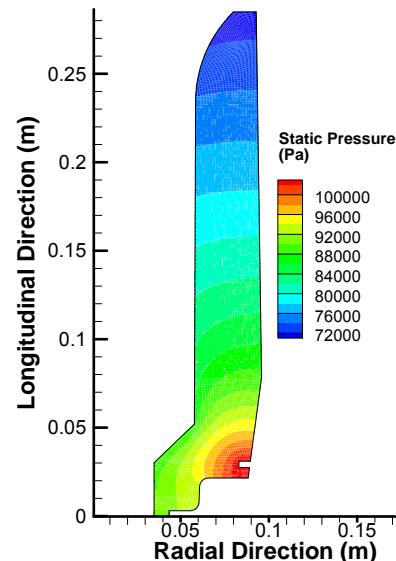


# Pressure Distribution

With/Without using Bernoulli's Equation for Inner Surface Pressure Estimation



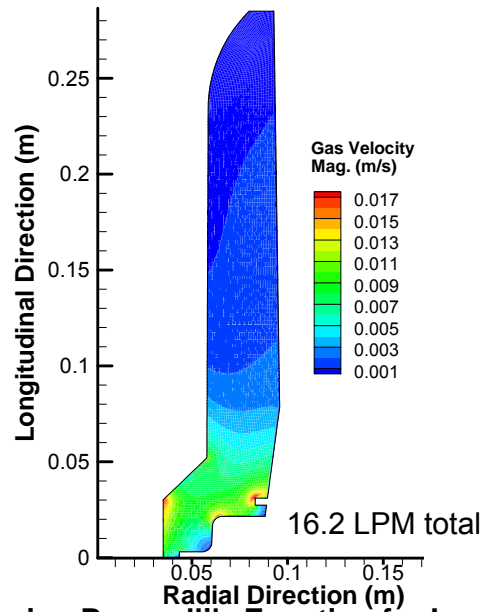
With Bernoulli's Equation for Inner Surface Pressure including hydrostatic & velocity terms



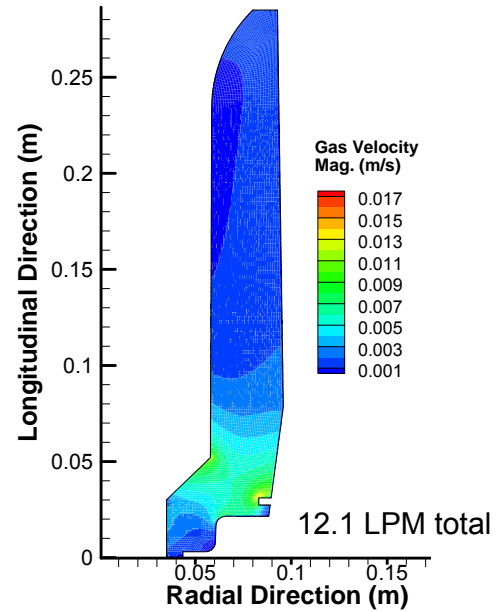
Hydrostatic Pressure only on Inner Surface

# Gas Velocity Distribution

With/Without steel velocity effect on inner surface pressure



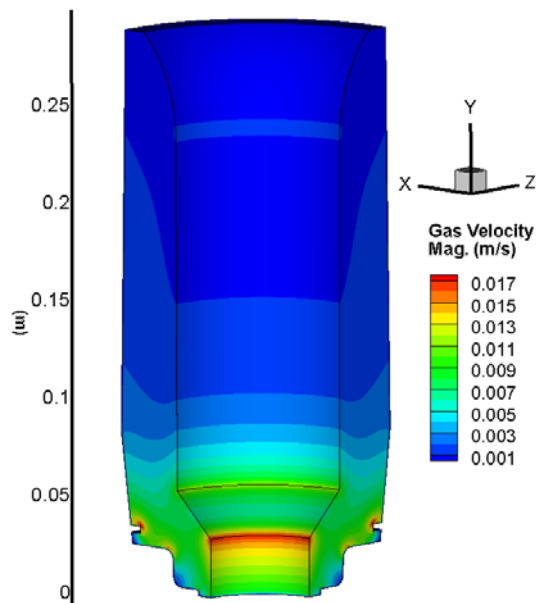
using Bernoulli's Equation for Inner Surface Pressure Estimation



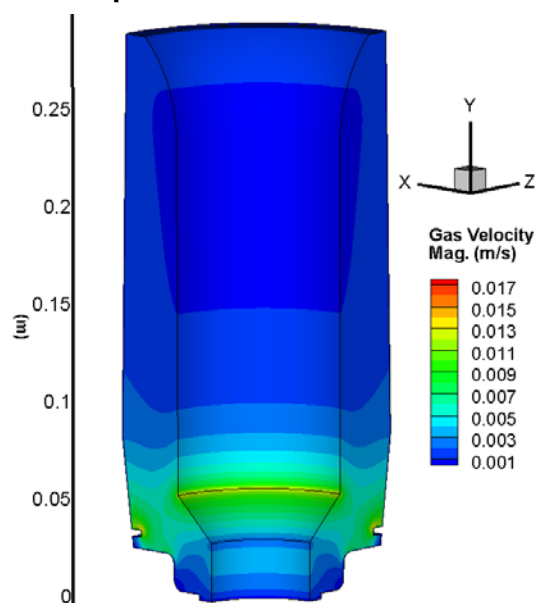
Without using Bernoulli's Equation

# Gas Velocity Distribution (3-D View)

using Bernoulli's Equation for Inner Surface Pressure Estimation



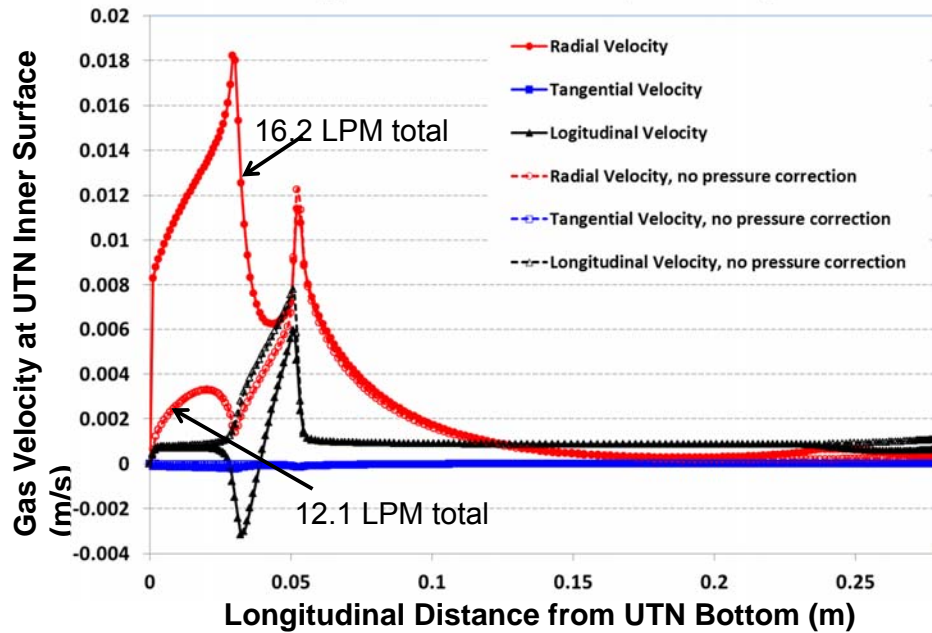
Without using Bernoulli's Equation





# Gas Flow Simulation

- Velocity Distribution (with/without pressure correction using Bernoulli's equation)



## Calculating Gas Flow Rate

$$\begin{aligned}
 pV &= nRT \\
 \frac{pV}{T} &= nR \\
 \frac{p_{hot} V_{hot}}{T_{hot}} &= nR = \frac{p_{std} V_{std}}{T_{std}} \\
 \frac{p_{hot} Q_{hot} \Delta t}{T_{hot}} &= \frac{p_{std} Q_{std} \Delta t}{T_{std}} \\
 \int_S \frac{p_{hot} Q_{hot} \Delta t}{T_{hot}} dS &= \int_S \frac{p_{std} Q_{std} \Delta t}{T_{std}} dS \\
 SQ_{std} &= \frac{T_{std}}{p_{std} T_{hot}} \int_S p_{hot} Q_{hot} dS \\
 \dot{V}_{total, std} &= 4SQ_{std} = 4 * 1.9156 \times 10^{-5} = 7.6624 \times 10^{-5} \text{ m}^3/\text{s}
 \end{aligned}$$

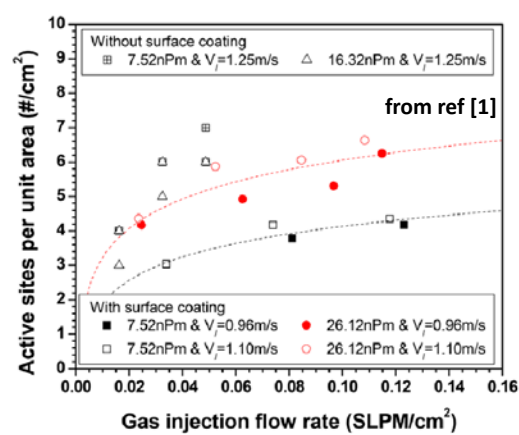
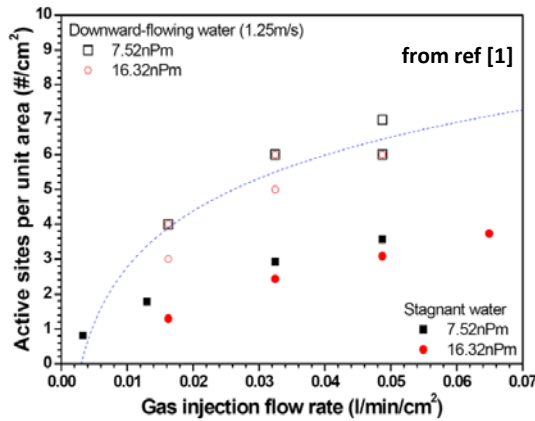
$T_{std}$ (K)	$T_{hot}$ (K)	$P_{std}$ (Pa)	$K_d$ (nmp)
300	1832	101325	10.1

$p_{hot}$  and  $Q_{hot}$  are the absolute pressure and gas hot flow rate at each of the face centers from the simulated data, then integrate them over the UTN inner surface to evaluate the gas volume flow rate in standard conditions as shown in the equation.

Since only a quarter is modeled, the total flow rate will be 4 times of the calculated value of  $S * Q_{std}$ .

In SLPM, it is:  $7.6624 * 10^{-5} * 1000 * 60 = 4.6 \text{ SLPM}$ , which matches reasonably well with measured gas flow rate of 5.01 SLPM. The mismatch could be caused by using a different permeability in the simulation, or possible gas leakage.

# Estimation of Active Sites Number at UTN Refractory Inner Surface



G. Lee and B.G. Thomas suggest<sup>[1]</sup>:

Active sites per unit area (#/cm<sup>2</sup>)

$$= 7 \times \frac{Q_g^{0.2635} \times U^{0.85} \times P_{erm}^{0.3308}}{\text{Radians}(\theta)}$$

Where:

$Q_g$ : the gas injection flow rate per cm<sup>2</sup> (LPM);

$U$ : liquid superficial velocity (m/s);

$P_{erm}$ : material permeability (npm);

$\theta$ : contact angle for wettability (rad)

Ref:

[1] G. Lee, B.G. Thomas, et al., Met. Mater. Int., Vol. 16, No. 3 (2010), pp. 501~506

So the number of active sites per cm<sup>2</sup> is:

$$N_{site} = \frac{7Q_g^{0.2635}U^{0.85}P_{erm}^{0.3308}}{\theta}$$

# Estimation of Mean Bubble Size using a Two-Stage Model

Hua Bai's two-stage initial bubble formation model<sup>[2]</sup>:

1. Expansion stage (solving for  $r$ , as  $r_e$ )

$$C_D \frac{1}{2} \rho_l \bar{u}^2 \pi r^2 = \frac{4}{3} \pi r^3 (\rho_l - \rho_g) g + \frac{1}{2} \pi r \sigma \sin \theta_o (\cos \theta_r - \cos \theta_a)$$

$$\text{Drag coefficient: } C_D = \frac{24}{\text{Re}_{bub}} (1 + 0.15 \text{Re}_{bub}^{0.687}) + 0.42 / (1 + 4.25 \times 10^4 \text{Re}_{bub}^{-1.16})$$

$$\text{Bubble Reynolds number: } \text{Re}_{bub} = \frac{\bar{u} D}{\nu}$$

$$\bar{u} = \frac{1}{2r} \int_{y=0}^{y=2r} u dy = 1.3173 U \frac{r^{1/7}}{D_N^{1/7}}$$

Based on the 1/7<sup>th</sup> law in turbulent flow in the circular pipe

$$\text{Contact angle function: } f_\theta = \sin \theta_o (\cos \theta_r - \cos \theta_a)$$

$$f_\theta(U) = -0.06079 + 0.33109U(\text{m/s}) + 0.078773U(\text{m/s})^2$$

Empirical correlation with liquid steel superficial velocity

2. Elongation stage (solving for  $r_d$ )

$$5.2692 \frac{\pi U}{Q_g D_N^{1/7}} \int_{r_e}^{r_d} \left( r^{15/7} (ar+b)^{3/2} + \frac{ar^{22/7}}{2} (ar+b)^{1/2} \right) dr = 2r_d e_d^{3/2} + \frac{d}{2} - r_e$$

$$a = \frac{e_d - 1}{r_d - r_e} \quad b = \frac{r_d - e_d r_e}{r_d - r_e} \quad e_d = 0.78592 + 0.70797U(\text{m/s}) - 0.12793U(\text{m/s})^2$$

$D_N$ : nozzle inner diameter (m)

$U$ : liquid steel superficial velocity (m/s)

$\rho_l$ : liquid density (kg/m<sup>3</sup>)

$\sigma$ : surface tension (N/m)

$\rho_g$ : gas density (kg/m<sup>3</sup>)

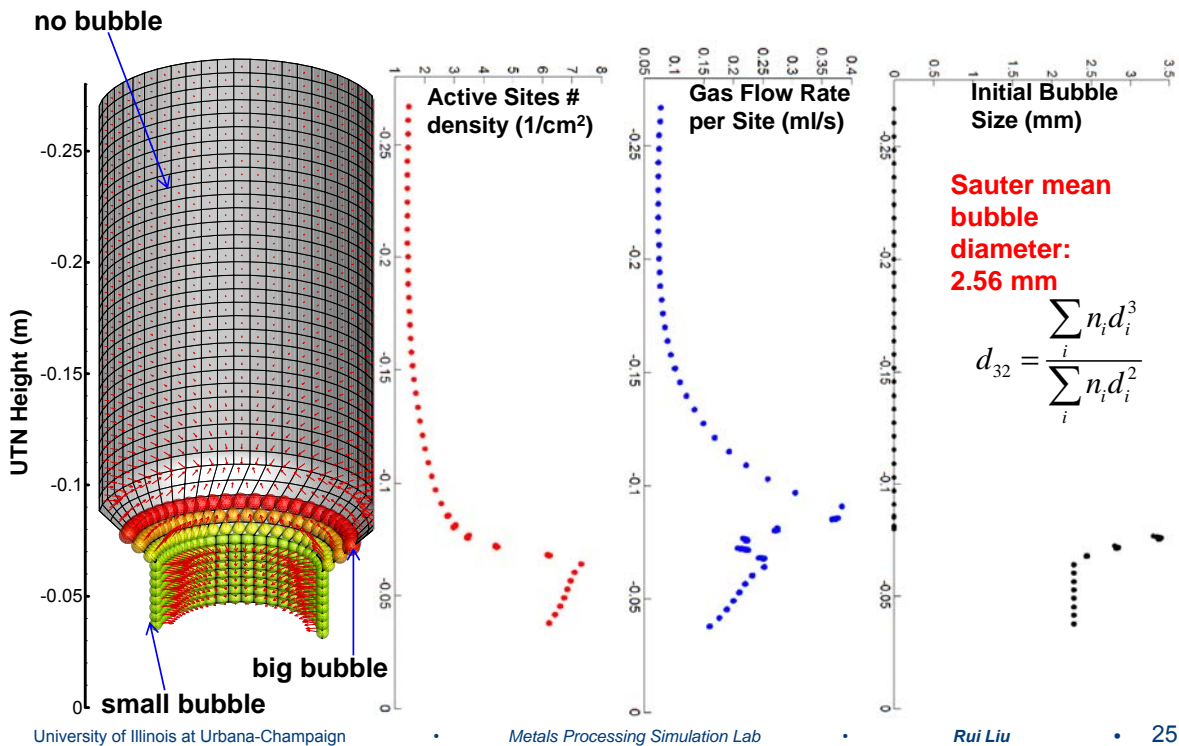
$\nu$ : kinematic viscosity of liquid steel (Pa\*s)

Ref:

[2] H. Bai and B. G. Thomas, Metall. Mater. Trans. B 32, 1143(2001).



# Initial Bubble Size Distribution at UTN Inner Surface



## PART 2 Conclusion

- Pressure at UTN inner surface serves as a boundary condition in the simulation of heated gas flow through porous refractory, thus needs careful treatment. Two different treatments of inner surface pressure were applied in current study:
  - Hydrostatic pressure only;
  - Hydrostatic pressure with correction using Bernoulli's equation (considering fluid kinetic energy change and pressure loss due to flow area change inside UTN)
- Results show:
  - Without pressure correction, hot argon flow rate is 12.1 LPM;
  - With pressure correction, hot argon flow rate is 16.2 LPM

## PART 2 Conclusion

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- A model system has been established to calculate hot argon flow rate at UTN inner surface, and estimate the active sites number density and eventually initial bubble size distributions.
- The initial bubble size is then used as an input parameter for multiphase flow simulations.
- For Severstal conditions for Nail Board 4,
  - Active sites number density ranges from 0 to  $\sim 7$  /cm<sup>2</sup>;
  - Initial bubble size ranges from  $\sim 2.2$  to  $\sim 3.5$  mm, with a Sauter-mean diameter = 2.56 mm;
  - Corresponding bubbling frequency ranges from  $\sim 40$  to  $\sim 180$  Hz.

## PART 3: Multi-phase Flow Simulations with Nail-Board Measurements

# Model Assumptions and Computational Details

- **Model assumptions:**
  - Isothermal flow
  - Argon-steel flow is within bubbly flow regime
  - No argon bubble break-up or coalescence, bubble size doesn't change with local pressure
  - Left-right symmetry for nozzle and mold domains
  - Relatively calm top surface, no significant gravity waves exist

## Governing Equations for Single-Phase Flow in Nozzle/Mold Region

- **Continuity Equation:**  $\nabla \cdot (\rho \mathbf{v}) = 0$
- **Momentum Conservation – Navier-Stoke Equation:**

$$\frac{\partial (\rho \mathbf{v})}{\partial t} + \nabla \cdot (\rho \mathbf{v} \mathbf{v}) = -\nabla p + \nabla \cdot ((\mu + \mu_T) \nabla \mathbf{v}) + \rho \mathbf{g}$$

- **Turbulence Model –  $k$ - $\omega$  SST URANS model [1]:**

**$k$  equation:** 
$$\frac{\partial k}{\partial t} + \mathbf{v} \cdot \nabla k = P_k - \beta^* k \omega + \nabla \cdot ((\nu + \sigma_k \nu_T) \nabla k)$$

**$\Omega$  equation:**

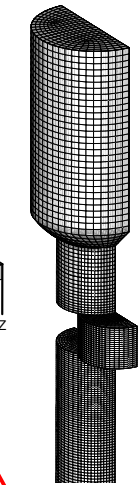
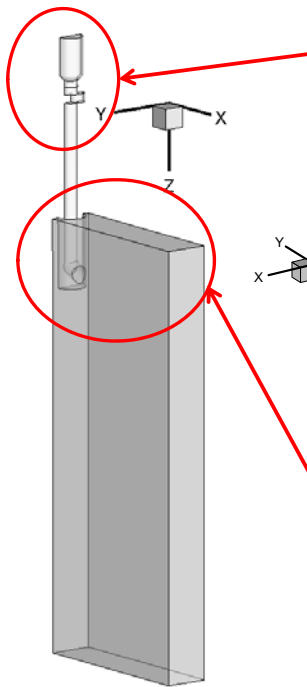
$$\frac{\partial \omega}{\partial t} + \mathbf{v} \cdot \nabla \omega = \alpha S^2 - \beta \omega^2 + \nabla \cdot ((\nu + \sigma_\omega \nu_T) \nabla \omega) + 2(1 - F_1) \sigma_{\omega 2} \frac{1}{\omega} (\nabla k \cdot \nabla \omega)$$

$$\nu_T = \frac{a_1 k}{\max(a_1 \omega, SF_2)} = \frac{\mu_T}{\rho}$$

**Values for the closure parameters can be found in [1].**

[1] F.R. Menter, "Two-Equation Eddy-Viscosity Turbulence Models for Engineering Applications", *AIAA Journal*, 1994, **32**(8), pp 1598

# Domain, Mesh and Methods for Fluid Flow in Severstal Caster Nozzle and Mold



Models and Schemes	Name
Turbulence Model	$k-\omega$ model
Multiphase Model	Eulerian Model
Advection Discretization	2 <sup>nd</sup> order upwinding scheme

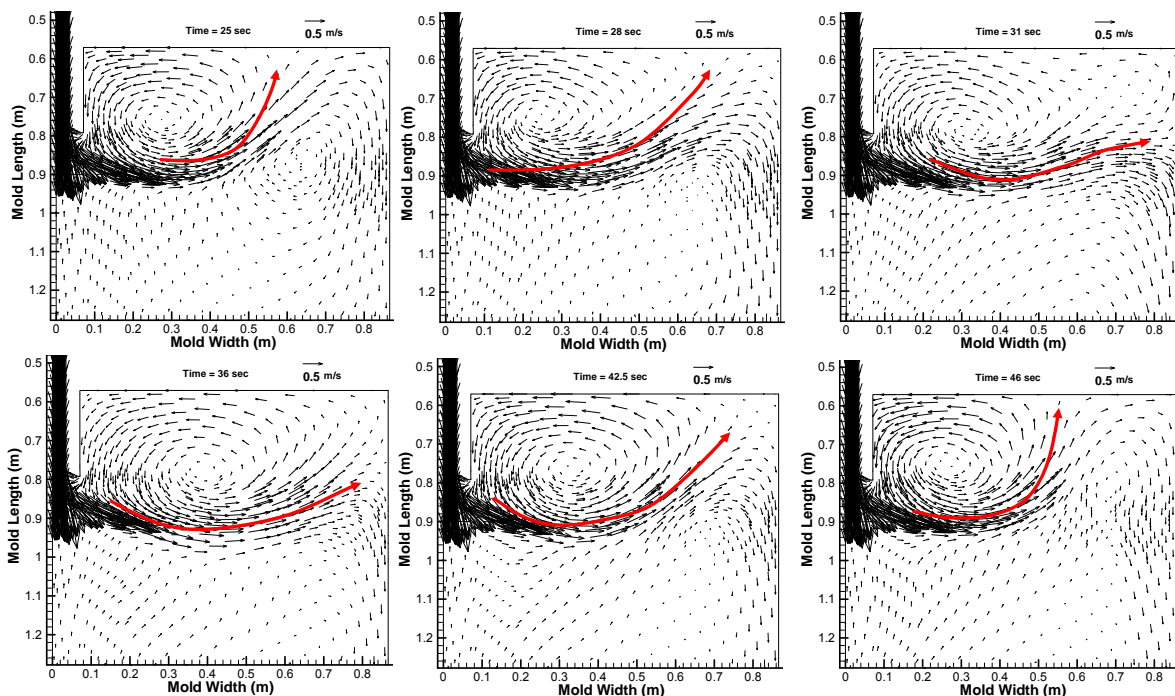
B.C.:

Meniscus	Domain Outlet
Free-slip wall with Gas Sink	Pressure outlet

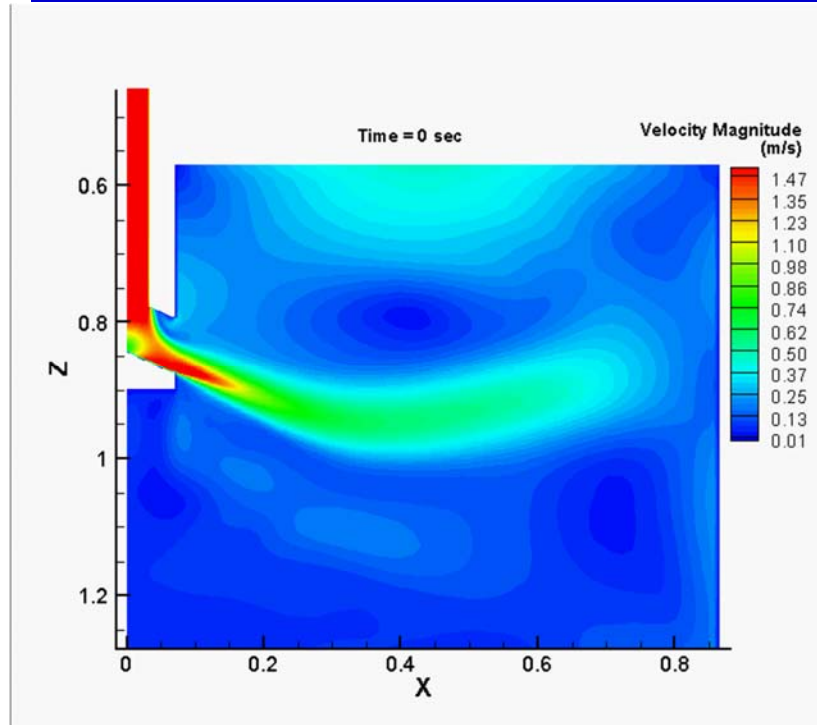
Time step:  
**0.05 sec**

Total mesh:  
**0.45 million mapped hexa-cells**

## BOARD 4: Single Phase Flow Pattern Evolution -- note jet wobble (~20s period)



# Single Phase Liquid Steel Flow Pattern -- Movie



## Multiphase Flow Models – Eulerian Bubble Model

- Eulerian-Eulerian Model, treat argon gas as another continuous phase

Continuity for argon:

$$\frac{\partial(\alpha_a \rho_a)}{\partial t} + \nabla \cdot (\alpha_a \rho_a \mathbf{v}_a) = 0$$

Continuity for steel:

$$\frac{\partial(\alpha_s \rho_s)}{\partial t} + \nabla \cdot (\alpha_s \rho_s \mathbf{v}_s) = 0$$

Momentum balance for argon:

$$\frac{\partial(\alpha_a \rho_a \mathbf{v}_a)}{\partial t} + \nabla \cdot (\alpha_a \rho_a \mathbf{v}_a \mathbf{v}_a) = -\alpha_a \nabla p + \nabla \cdot (\alpha_a \mu_a \nabla \mathbf{v}_a) + K_{as} (\mathbf{v}_s - \mathbf{v}_a) + \alpha_a \rho_a \mathbf{g}$$

Momentum balance for liquid steel:

$$\frac{\partial(\alpha_s \rho_s \mathbf{v}_s)}{\partial t} + \nabla \cdot (\alpha_s \rho_s \mathbf{v}_s \mathbf{v}_s) = -\alpha_s \nabla p + \nabla \cdot (\alpha_s (\mu_s + \mu_t) \nabla \mathbf{v}_s) + K_{as} (\mathbf{v}_a - \mathbf{v}_s) + \alpha_s \rho_s \mathbf{g}$$

$$K_{as} = \frac{3}{4} \frac{C_D}{D_b} \alpha_s \rho_s |\mathbf{v}_s - \mathbf{v}_a|, \quad C_D = \frac{24}{\text{Re}_b} (1 + 0.15 \text{Re}_b^{0.687}), \quad \text{Re}_b = \frac{\rho_s |\mathbf{v}_s - \mathbf{v}_a| D_b}{\mu_s}$$

$$\mu_t = C_\mu \rho_s \frac{k^2}{\varepsilon}$$

Volume fraction equation:

$$\alpha_s + \alpha_a = 1$$

# Another Multiphase Flow Model – Discrete Phase Bubble Model

- Lagrangian bubble tracking, calculate the bubble trajectories

$$m_p \frac{d\mathbf{v}_p}{dt} = \mathbf{F}_D + \mathbf{F}_L + \mathbf{F}_{added-mass} + \mathbf{F}_G + \mathbf{F}_{press} + \mathbf{F}_{stress}$$

Drag force
Lift force
Added mass force
Gravity
Pressure & Stress gradient force

**Motion:**  $\mathbf{v}_p = \frac{d\mathbf{x}_p}{dt}$

$$\mathbf{F}_D = \frac{1}{8} \pi d_p^2 \rho_f C_D |\mathbf{v}_f - \mathbf{v}_p| (\mathbf{v}_f - \mathbf{v}_p)$$

$$C_D = f_{Re_p} \left( \frac{24}{Re_p} \right), Re_p = \left| (\mathbf{v}_f - \mathbf{v}_p) \frac{d_p}{\nu} \right|$$

$$f_{Re_p} = (1 + 0.15 Re_p^{0.687})$$

$$\mathbf{F}_G = \frac{\pi d_p^3}{6} \rho_p \mathbf{g}$$

$$\mathbf{F}_{press} + \mathbf{F}_{stress} = \frac{\rho_f \pi d_p^3}{6} \left( \frac{D\mathbf{v}_f}{Dt} - \mathbf{g} \right)$$

$$\mathbf{F}_{added-mass} = \frac{\rho_p \pi d_p^3}{12} \left( \frac{d\mathbf{v}_f}{dt} - \frac{d\mathbf{v}_p}{dt} \right)$$

$$\mathbf{F}_L = -\frac{9}{4\pi} \mu d_p^2 \mathbf{U}_s \operatorname{sgn}(G) \left[ \frac{|G|}{\nu} \right]^{\frac{1}{2}} J^u$$

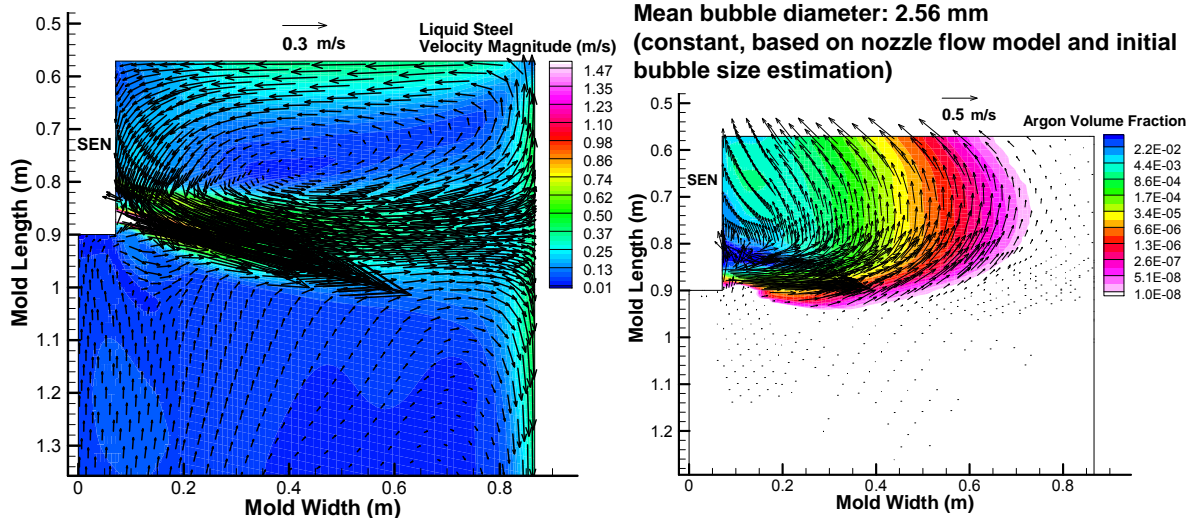
$$G = \mathbf{U}_s \times (\nabla \times \mathbf{v}_f), \quad \nabla \varepsilon = \operatorname{sgn}(G) \frac{\sqrt{|G|} \nu}{U_s}, \quad \mathbf{U}_s = \mathbf{v}_p - \mathbf{v}_f$$

$$J(\varepsilon) = 0.6765 * (1 + \tanh[2.5 \lg \varepsilon + 0.191]) (0.667 + \tanh[6(\varepsilon - 0.32)])$$

**Ref:**  
 (1) Maxey, M.R. and Riley, J.J.: Physics of Fluids, 1983, vol. 26 (4), pp. 883-889.  
 (2) Crowe, C., Sommerfeld, M. and Tsinji, Y.: Multiphase Flows with Droplets and Particles, CRC Press, 1998, pp. 23-95.

## Typical Argon-Steel Flow Pattern in SEN/Mold Region

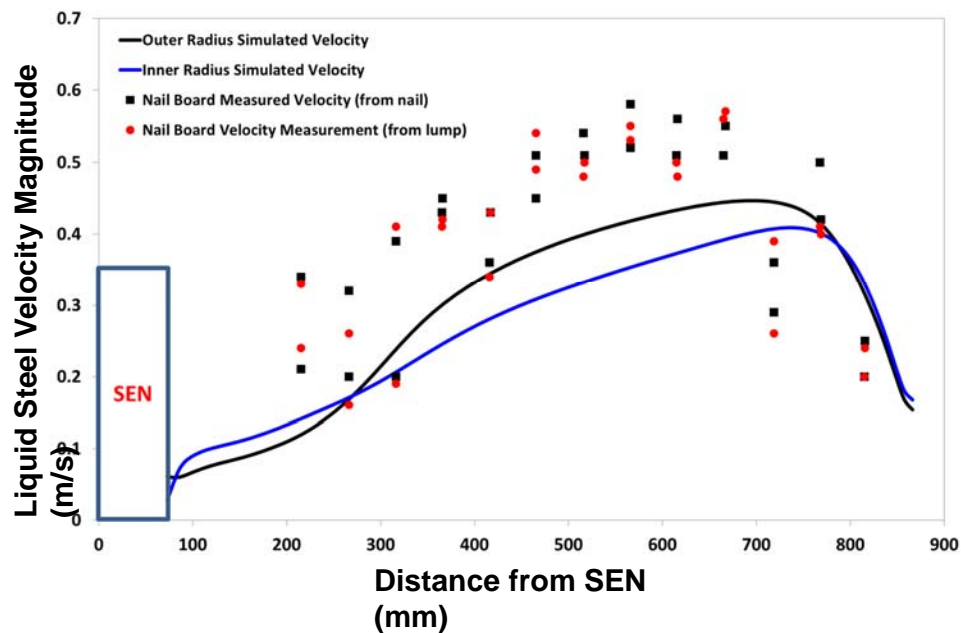
- Eulerian model simulation results:



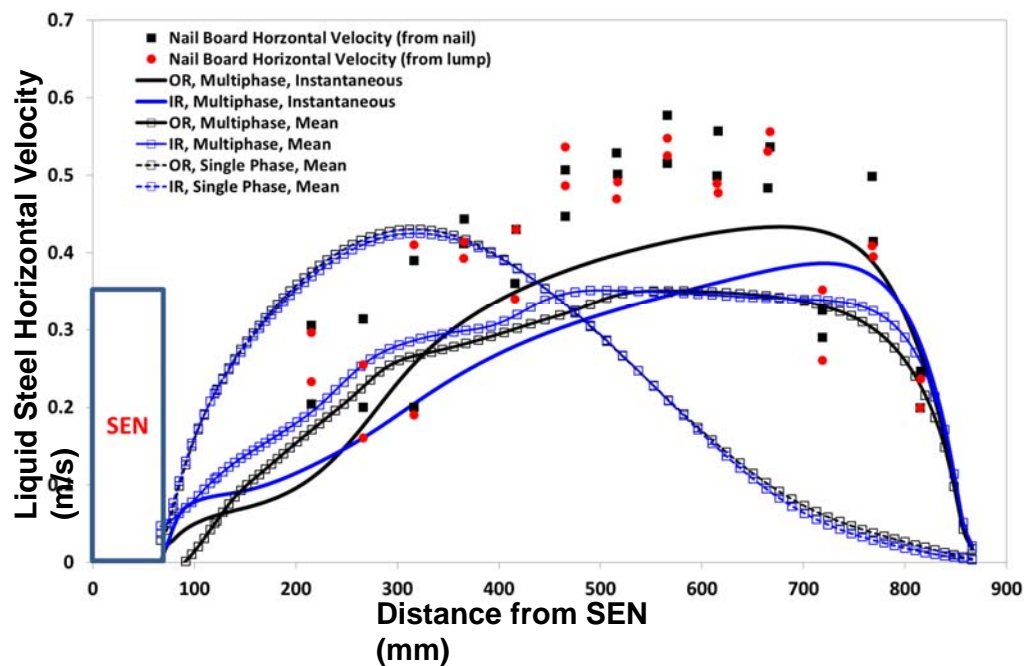
- Double-roll flow pattern still exists with argon injection (~4% argon)
- Large scale instability (such as jet wobbling) has been reduced by gas injection comparing with single phase flow pattern in this case



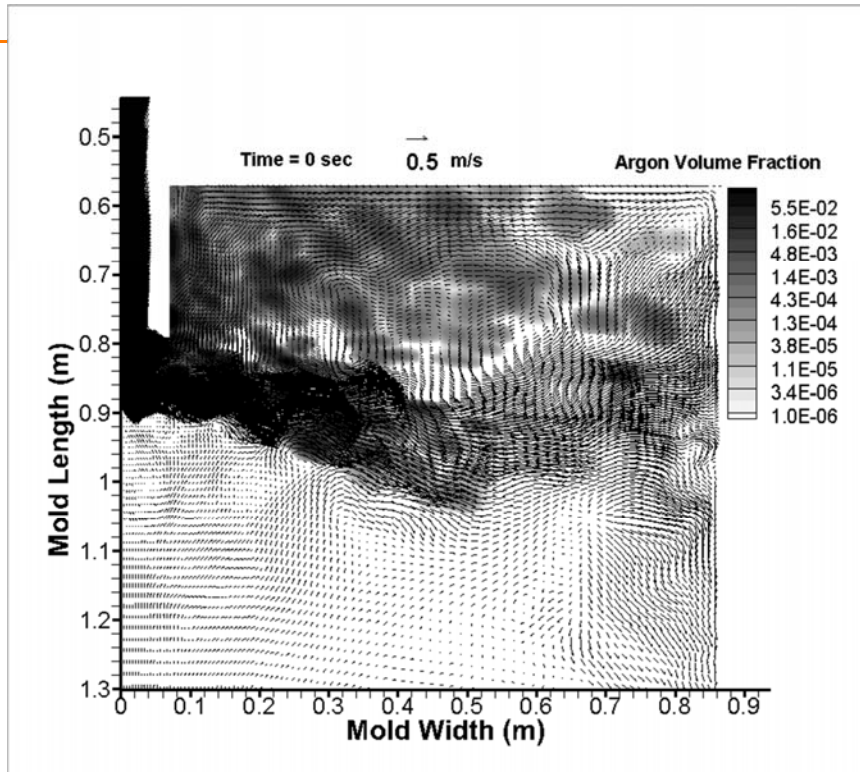
# Comparison with Nail-Board Experiment Results – Board 4 Velocity Magnitude



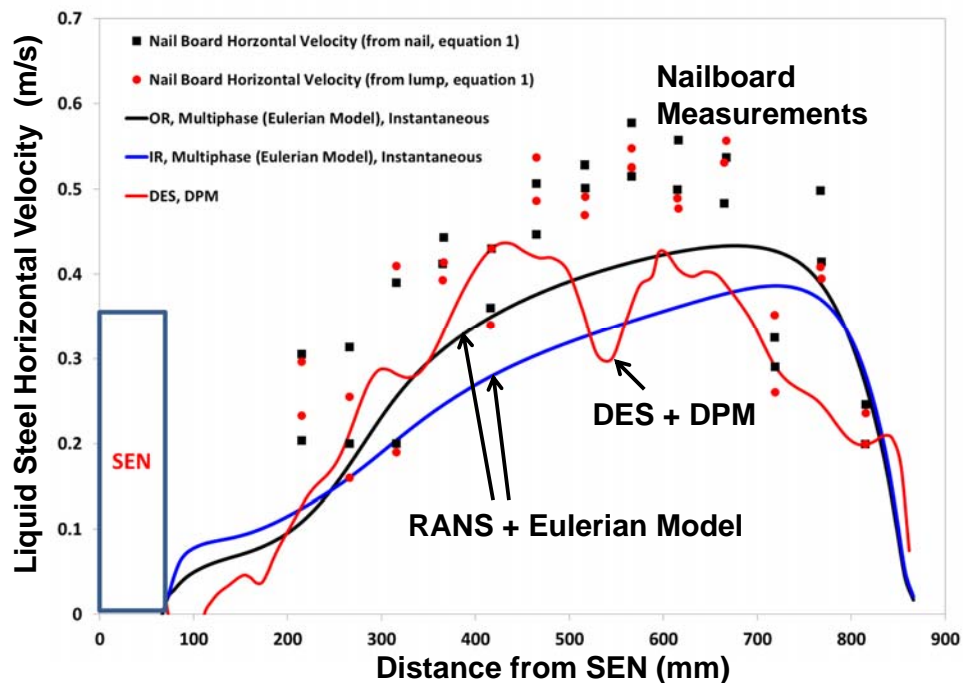
# Comparison of Horizontal Velocity Profiles



## BOARD 4: DES with DPM

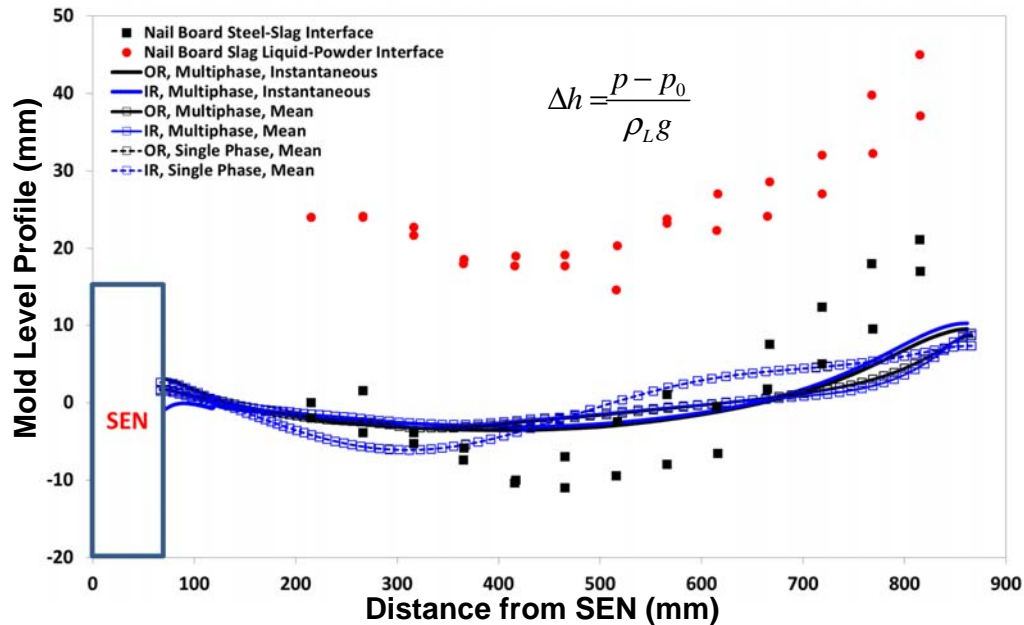


## Comparison of Horizontal Velocity Profiles – Model vs. Measurements (Board 4)





# Comparison of Mold Level Profile (BOARD 4)



## Conclusion – from BOARD 4 Simulation

- Single phase flow simulation results show a wobbling liquid steel jet, which causes a periodical reverse flow away from SEN at regions close to the narrow face;
- Jet wobbling is reduced by adding a small amount of gas into the liquid steel (~4% in current case, board 4), and the double-roll flow pattern still remains;
- Comparison of liquid steel surface velocity with nail board measurements suggests:
  - the trend of both instantaneous and mean steel surface velocity distributions from argon-steel two-phase simulations match well with nail-board measurements (with velocities peak closer to narrow face), but with a ~20-30% velocity magnitude difference;
  - single phase simulation results suggest an opposite velocity distribution – velocity peaks close to SEN instead of narrow face;

## Conclusion – from BOARD 4 Simulation (CONT.)

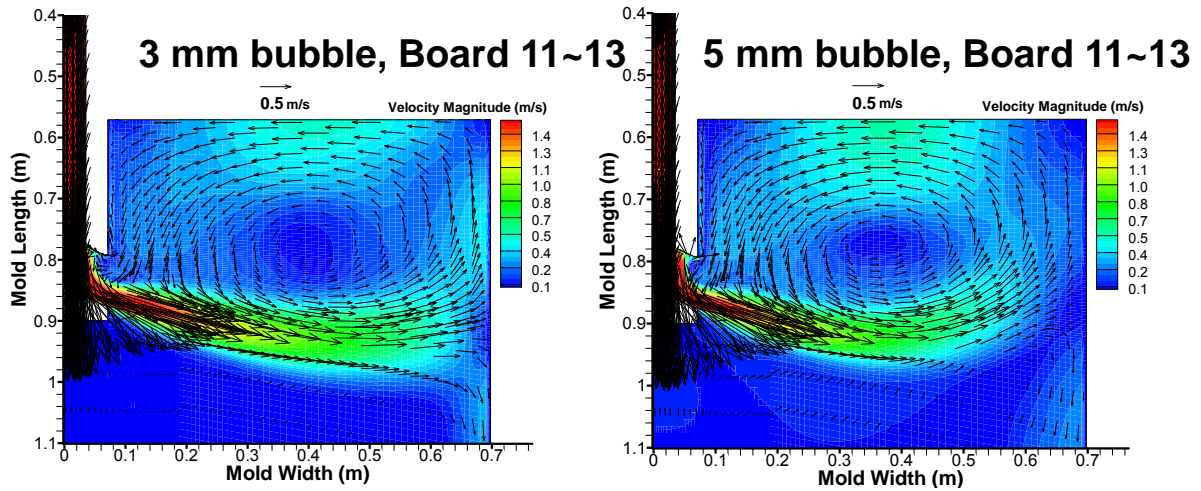
- **DPM model is promising in multiphase flow modeling, given:**
  - A high resolution mesh
  - More accurate turbulence model (LES e.g.)
  - A well-educated bubble size distribution

## Simulations for BOARD 11~13

Date & Time	Nailboard Case #	Casting Speed (inch/min)	Strand Width (inch)	Argon Flow Rate (SLPM)	Argon Back Pressure (psi)	Submerging Depth (mm)
10/16, 3:28:54 pm	11	65.0	54.99	7.0	18.07	222
10/16, 3:29:20 pm	12	65.0	54.99	7.0	18.07	222
10/16, 3:29:41 pm	13	65.0	54.99	7.0	18.07	222

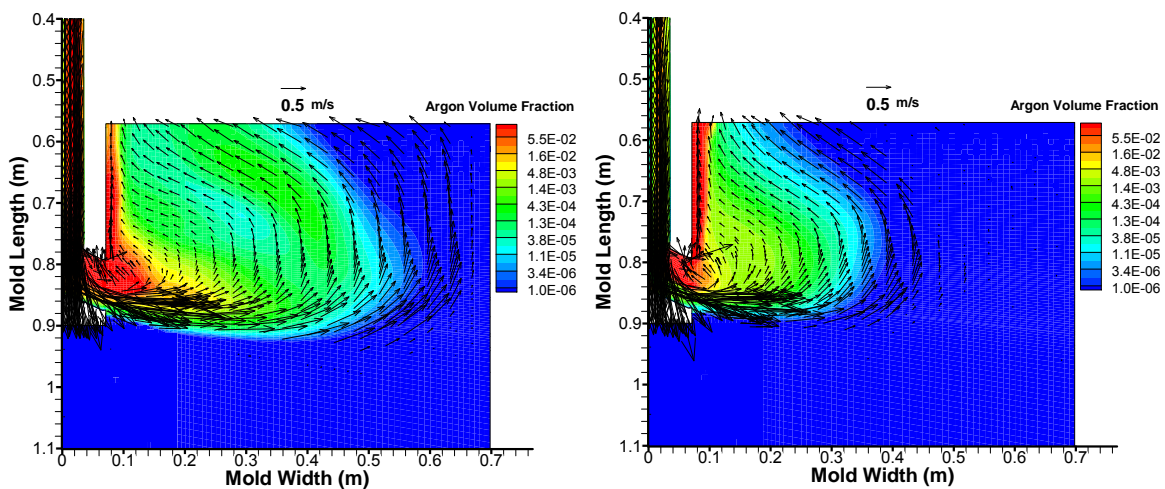
- **In this part of work, the following techniques and parameters are utilized:**
  - RANS model (k-epsilon) for turbulence
  - Eulerian model for dispersed bubble phase
  - Two mean bubble diameters, 3mm and 5mm, are studied
  - Mold top surface profile is calculated using:
    - Pressure method
    - A moving-grid free surface tracking algorithm

# Liquid Steel Velocity Distribution at Mold Center Plane



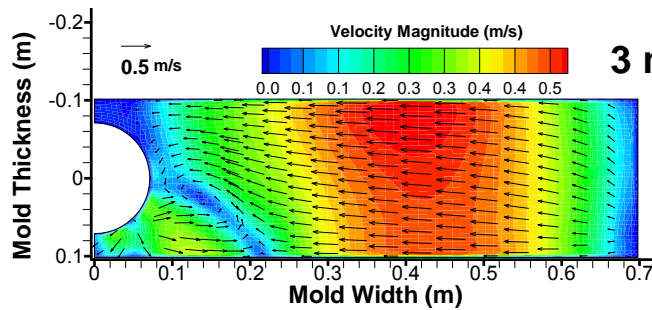
- Double-roll flow pattern achieved for both 3 mm bubble and 5 mm bubble
- Slightly higher sub-meniscus velocity is found for 5 mm bubble size case

# Argon Gas Velocity / Fraction Distribution at Mold Center Plane

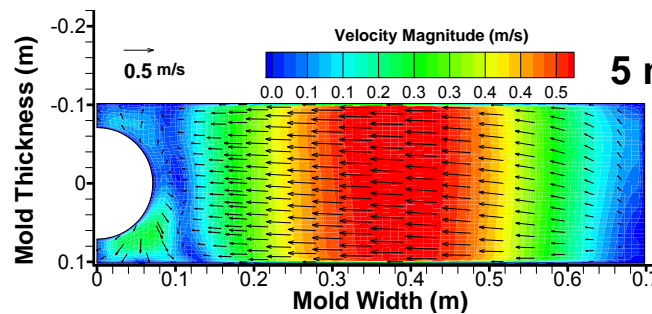


- Gas with 3 mm bubble size spreads further into the liquid pool, closer to the narrow phase, comparing with the case with 5 mm bubble size

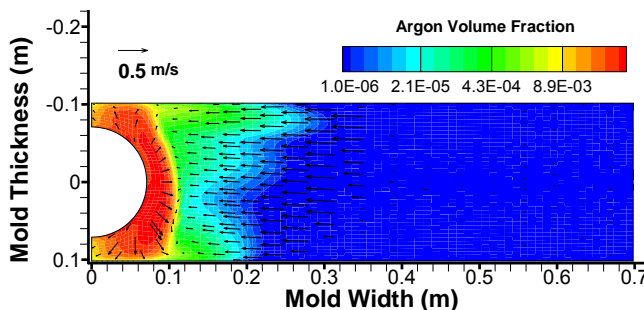
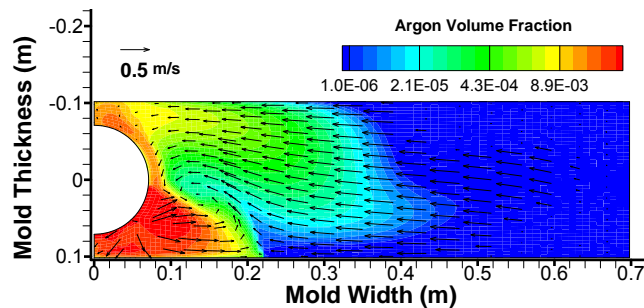
# Liquid Steel Velocity Distribution at Mold Top Surface



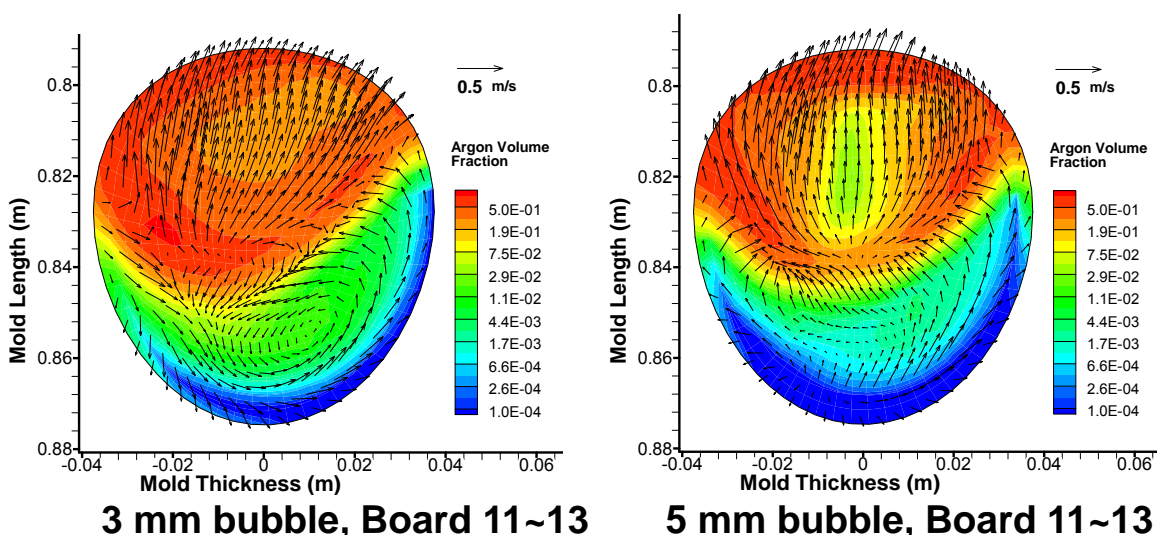
- Higher surface velocity found in 5 mm bubble size case



# Argon Gas Velocity / Fraction Distribution at Mold Top Surface

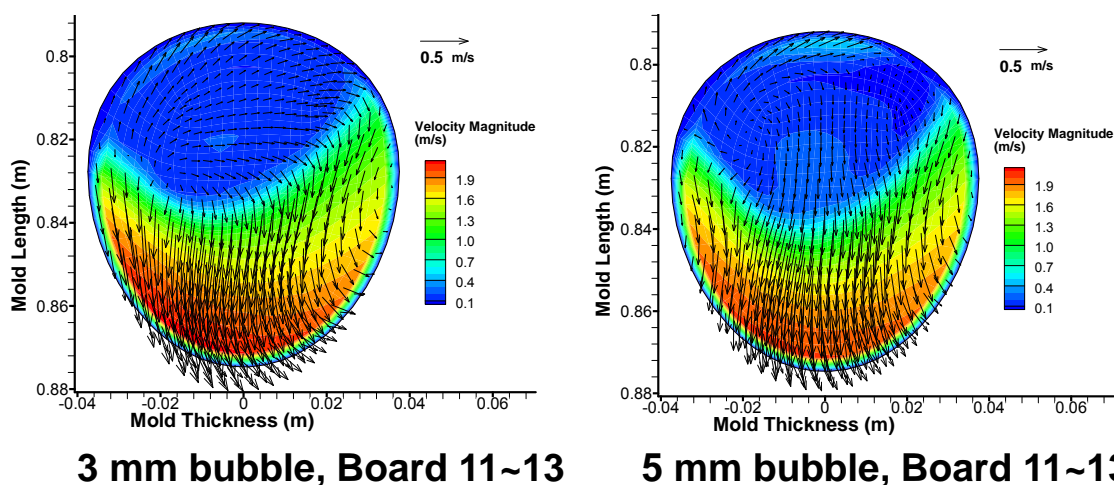


# Argon Gas Velocity / Fraction Distribution at Port Exit



- Higher gas rising velocity found in 5 mm bubble size case at port exit

# Liquid Steel Velocity Distribution at Port Exit





# Free Surface Modeling

- **Pressure Conversion**
  - Wall boundary condition at domain top surface
  - Best used for steady / quasi-steady state flows, without gravity waves
    - Lifting
    - displacement
- **Moving-grid Surface Tracking \***
  - Pressure boundary at domain top surface
  - Working for both steady state and transient flows, with/without gravity waves

\* R. Liu, B.G. Thomas, L. Kalra, T. Bhattacharya, and A. Dasgupta., *Proc. AISTech 2013 Conf.* (Pittsburg, PA), p1351-1364, (2013)

## Moving Grid Technique using FVM

Continuity equation with moving mesh:

$$\nabla \cdot (\rho(\mathbf{v} - \mathbf{v}_g)) = 0$$

Momentum equation with moving mesh:

$$\frac{\partial(\rho \mathbf{v})}{\partial t} + \nabla \cdot (\rho \mathbf{v}(\mathbf{v} - \mathbf{v}_g)) = -\nabla p + \nabla \cdot (\mu \nabla \mathbf{v}) + \mathbf{F}$$

$\mathbf{v}$  is fluid velocity, while  $\mathbf{v}_g$  is grid velocity (mesh velocity)

Kinematic B.C.:

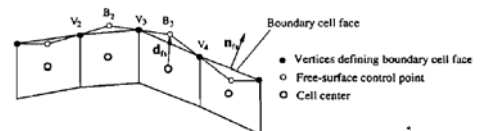
$$[(\mathbf{v} - \mathbf{v}_s) \cdot \mathbf{n}]_{fs} = 0 \quad \text{or} \quad \dot{m}_{fs} = 0$$

Dynamic B.C.: all forces in equilibrium at fs.

This node moving approach has been adopted and coded into FLUENT udf for free surface modeling in current work

Ref:

S. Muzaferija and M. Perić, Numerical Heat Transfer, Part B: Fundamentals: An International Journal of Computation and Methodology, 1997. Vol 32:4, 369-384



$$\dot{m}_{fs} + \rho \dot{V}'_{fs} = 0$$

$$\mathbf{r}_{B_i}^{k+1} = \mathbf{r}_{B_i}^k + \Delta h \mathbf{e}_{fs}$$

$$\mathbf{r}_{V_i}^{k+1} = \mathbf{r}_{V_i}^k - \mathbf{e}_{fs} \cdot \left[ \mathbf{r}_{V_i}^k - \sum_{m=1}^n w_m \mathbf{r}_{B_m}^{k+1} \right] \mathbf{e}_{fs}$$

$$\Delta h = \gamma_{fs} \frac{\dot{V}'_{fs} \Delta t}{S_{fs} \mathbf{n} \cdot \mathbf{e}_{fs}}$$

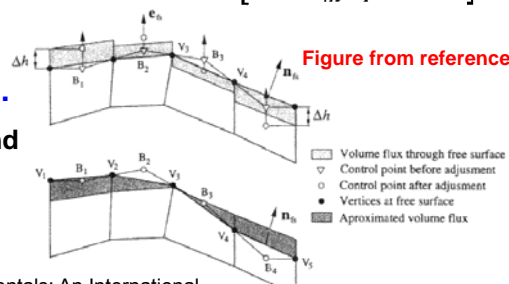
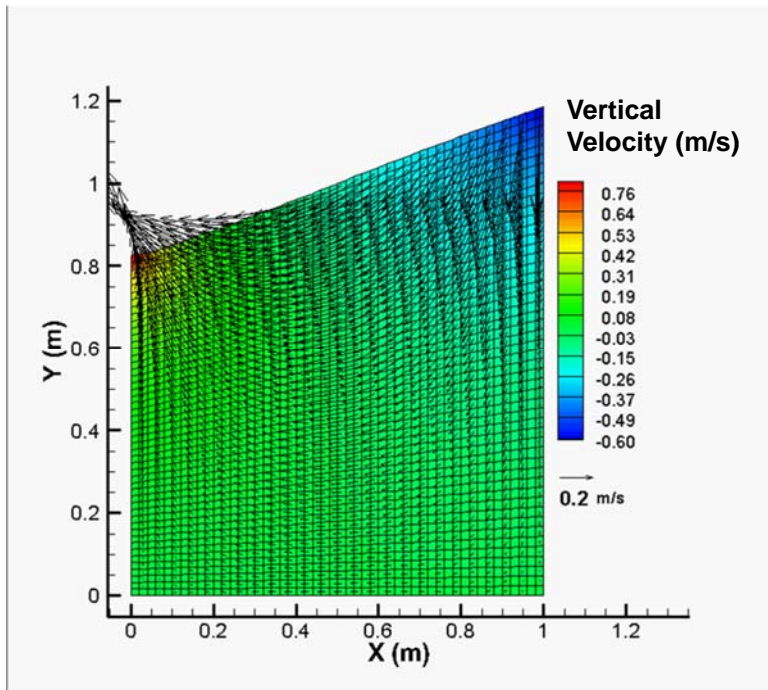


Figure from reference

# Large Amplitude Sloshing



- Free surface nodes moved via UDF, with kinematic and dynamic B.C.s satisfied
- Side wall nodes and internal nodes are smoothed via solution of a Laplace equation diffusing free surface nodes displacements to interior nodes

## Model Validation – Analytical Solution and Case Setup

- 2-D small-amplitude sloshing problem

$$a(t) = \frac{4\nu^2 k^4}{8\nu^2 k^4 + \omega_0^2} a_0 \operatorname{erfc}(\nu k^2 t)^{\frac{1}{2}} + \sum_{i=1}^4 \frac{z_i}{Z_i} \left( \frac{\omega_0^2 a_0}{z_i^2 - \nu k^2} \right) \exp((z_i^2 - \nu k^2)t) \operatorname{erfc}\left(z_i t^{\frac{1}{2}}\right)$$

Where  $z_i$  is the  $i^{\text{th}}$  root of the equation below, and  $z_i$  is defined as:

$$z^4 + 2\nu k^2 z^2 + 4(\nu k^2)^{\frac{3}{2}} z + \nu^2 k^4 + \omega_0^2 = 0 \quad \omega_0 = \sqrt{gk}$$

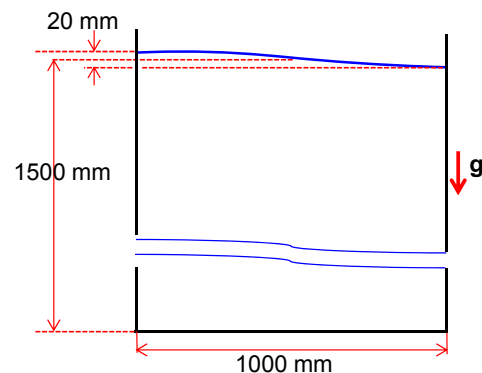
Prosperetti, A., 1981. "Motion of two superposed viscous fluids". Physics of Fluids, 24(7), July, pp. 1217–1223.

Initial Interface:  $h(x) = 1.5 + a_0 \sin(\pi(0.5 - x))$

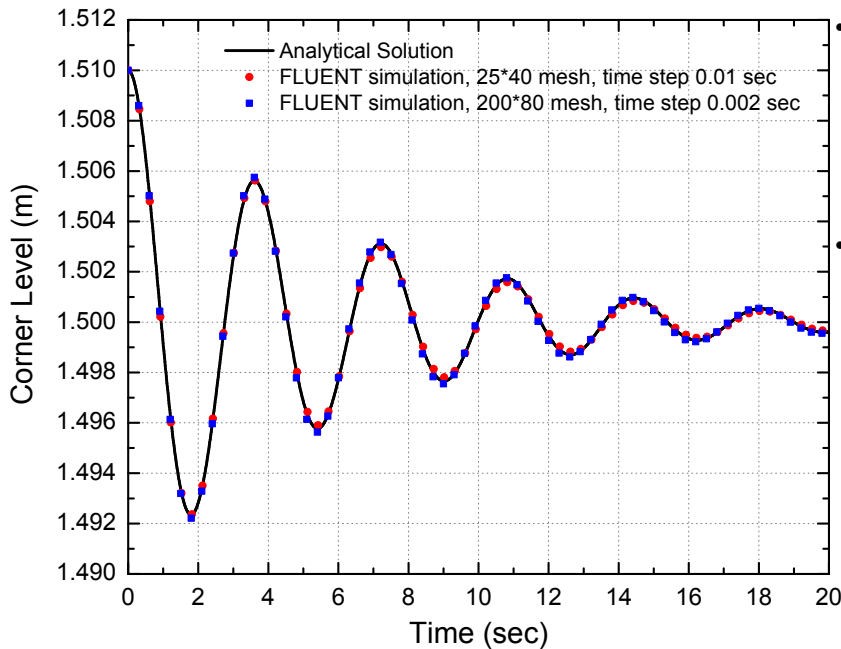
$$a_0 = 0.01 \text{ m}$$

Level Height (mm)	Tank Width (mm)	Initial Perturbation Amplitude (mm)
1500	1000	20

Fluid Kinematic Viscosity (m <sup>2</sup> /s)	Gravity Acceleration (m/s <sup>2</sup> )	Surface Tension (N/m)
0.01	1.0	0

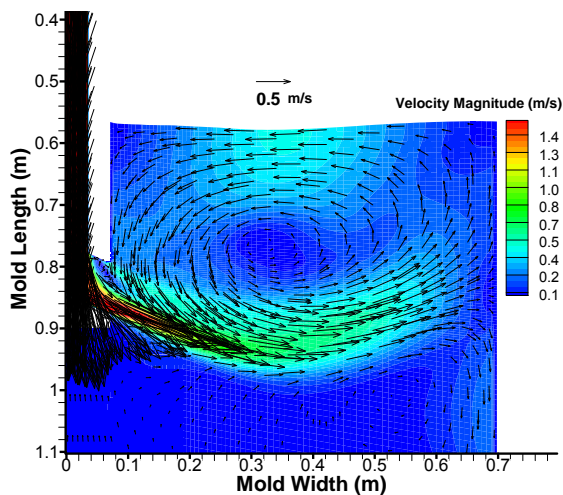


# Model Validation – Comparison with Analytical Solution

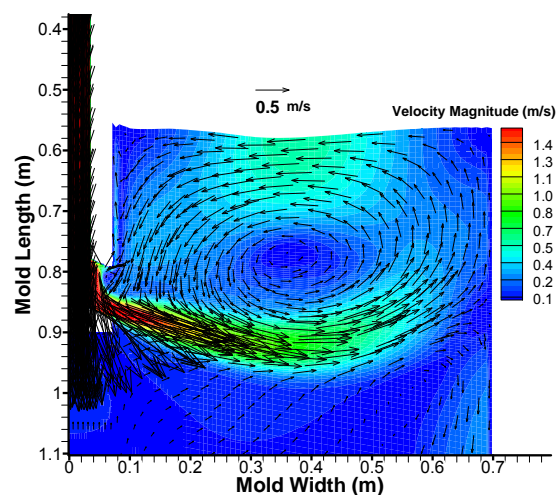


- Excellent match with analytical solution obtained even for simulations using the very coarse mesh
- Using second order (or higher) advection scheme and temporal scheme are crucial for achieving accuracy

## Liquid Steel Velocity Distribution at Mold Center Plane (Board 11-13)



Single Phase, Board 11~13

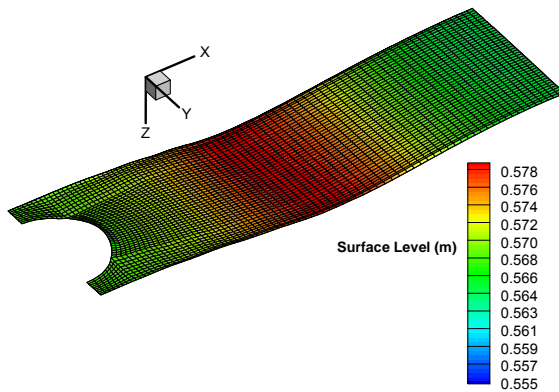


Multiphase, Board 11~13  
(5 mm bubble size)

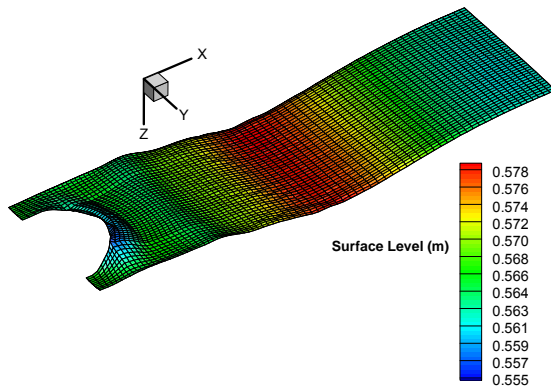
- Gas injection (with 5 mm bubble size) increases liquid steel surface velocity, and surface level difference



# Moving-Grid Free Surface Tracking Single Phase vs. Two-Phase Flow



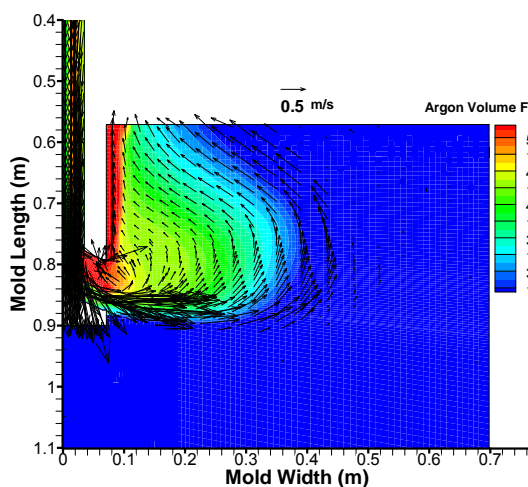
Single Phase, Board 11~13



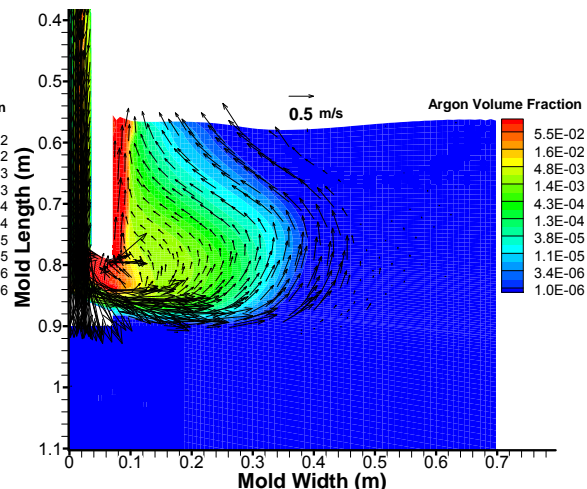
Multiphase, Board 11~13  
(5 mm bubble size)

- Rising of gas bubbles elevate the surface level near SEN, comparing with the no-gas case

# Liquid Steel Velocity Distribution – Wall B.C. vs. Pressure B.C.



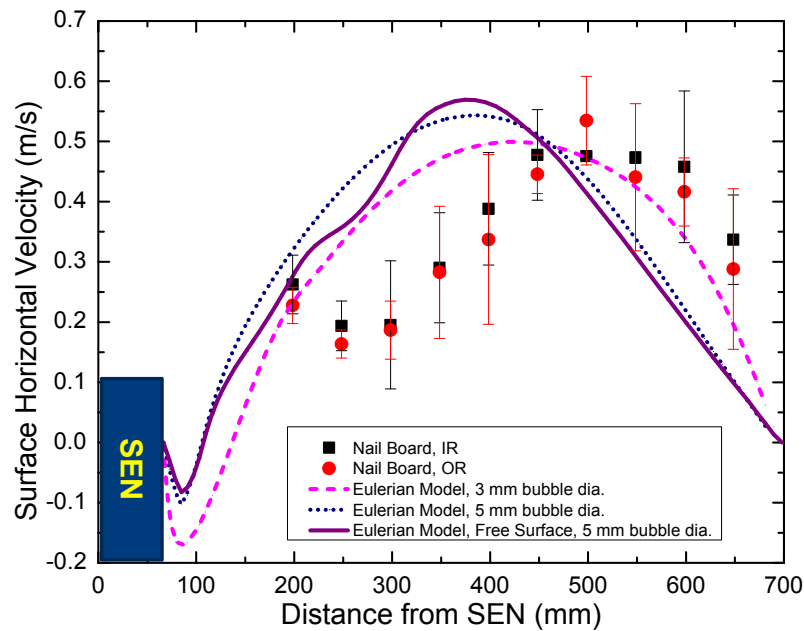
5 mm bubble, Board 11~13  
(wall B.C. on top)



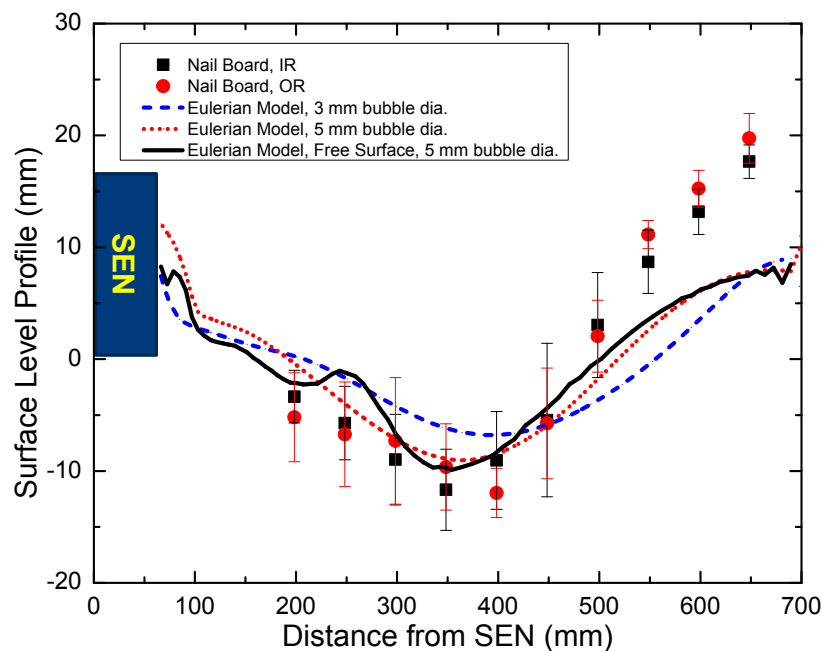
5 mm bubble, Board 11~13  
(free surface tracking, pressure  
B.C. on top)

- No significant change for the gas distribution using either method

# Comparison with Nail Board Measurements – Horizontal Velocity



# Comparison with Nail Board Measurements – Surface Level Profile



## Conclusions – from BOARD 11~13

- Single phase flow simulation results does not show any wobbling liquid steel jet, thus narrower mold generates a more stable flow pattern comparing to the wider mold (shown previously in case 4)
- Smaller bubble size (3 mm) increases surface velocity slightly, while larger bubbles (5 mm)
  - increase the surface velocities significantly and
  - increase surface level differences
- Simple pressure method and moving-grid method predict similar surface level profiles
- The 5-mm bubble case matches reasonably with measurements (better than with 3mm dia.)
- Better match near NF could be obtained if the nailboard had been tilted

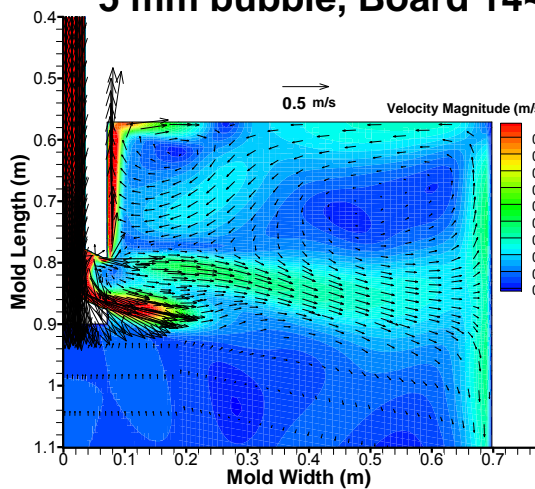
## Simulations for BOARD 14~16

Date & Time	Nailboard Case #	Casting Speed (inch/min)	Strand Width (inch)	Argon Flow Rate (SLPM)	Argon Back Pressure (psi)	Submergence Depth (mm)
10/16, 3:35:50 pm	14	25.5	54.99	6.3	19.18	222
10/16, 3:36:16 pm	15	25.5	54.99	6.2	19.18	222
10/16, 3:36:36 pm	16	25.5	54.99	6.2	19.18	222

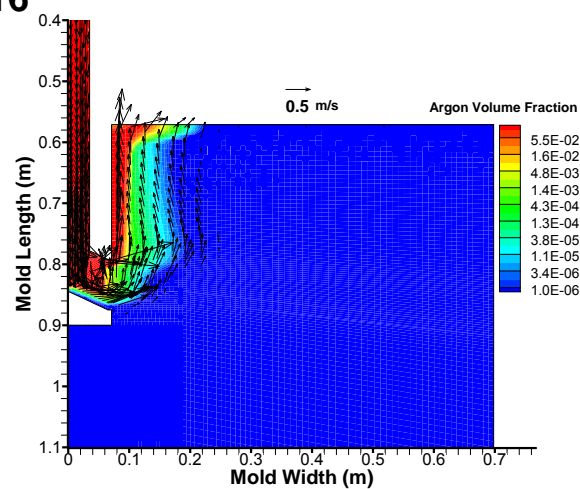
- In this part of work, the following techniques and parameters are utilized:
  - RANS model (k-epsilon) for turbulence
  - Eulerian model for dispersed bubble phase
  - Three mean bubble diameters, 3mm, 5mm and 8mm, are studied

# Liquid Steel Velocity Distribution at Mold Center Plane – Board 14~16

5 mm bubble, Board 14~16

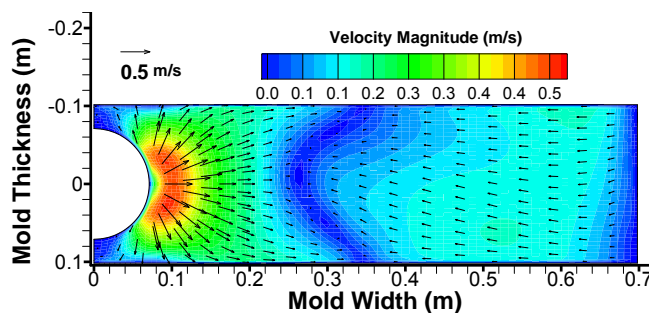


Complex flow pattern with  
partial reversed flow

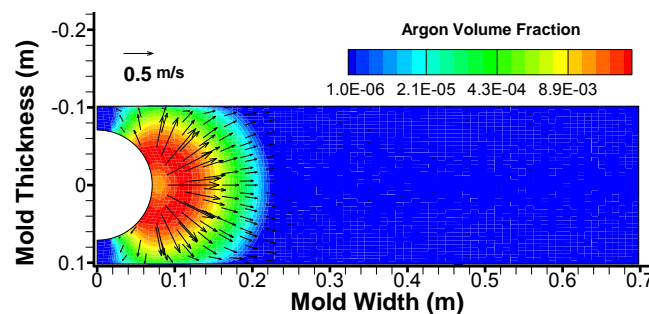


Rising argon bubbles  
concentrate near SEN

## Velocity / Argon Fraction Distribution at Mold Top Surface – Board 14~16

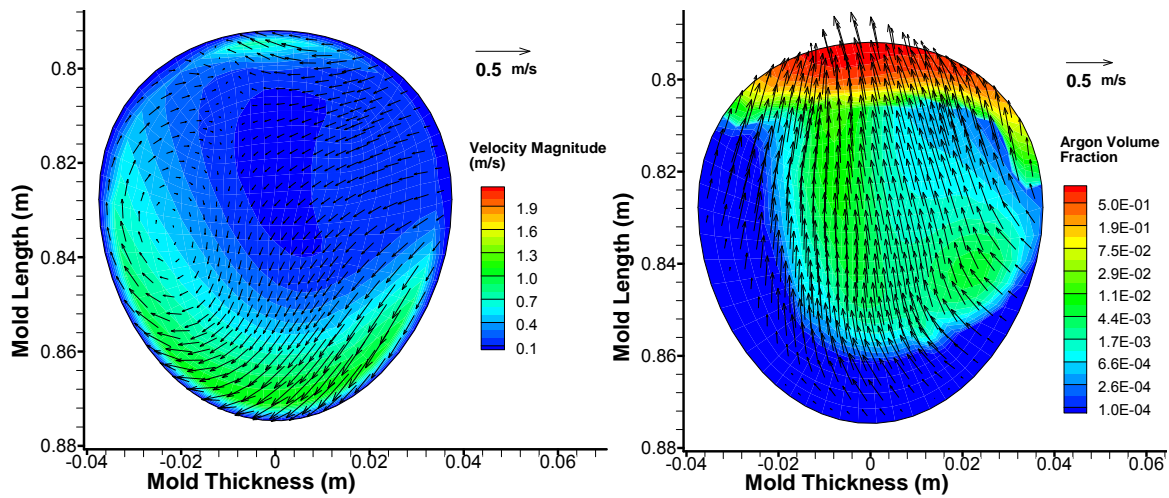


- Two liquid steel surface streams move in opposite directions, and meet in the middle region

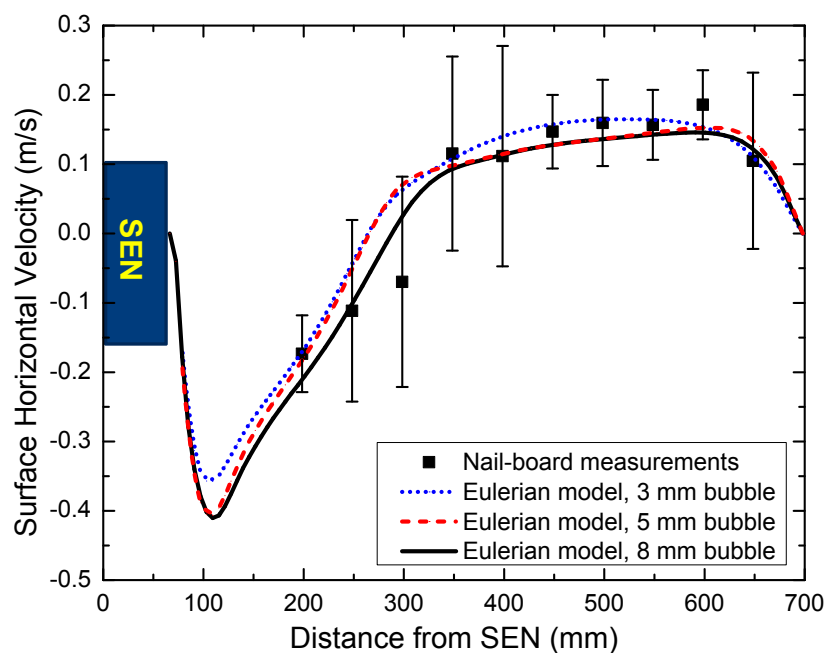


- Gas exits to the top surface in a symmetric manner, indicating less affected by the swirling liquid steel jet

# Liquid Steel Velocity / Gas Distribution at Port Exit

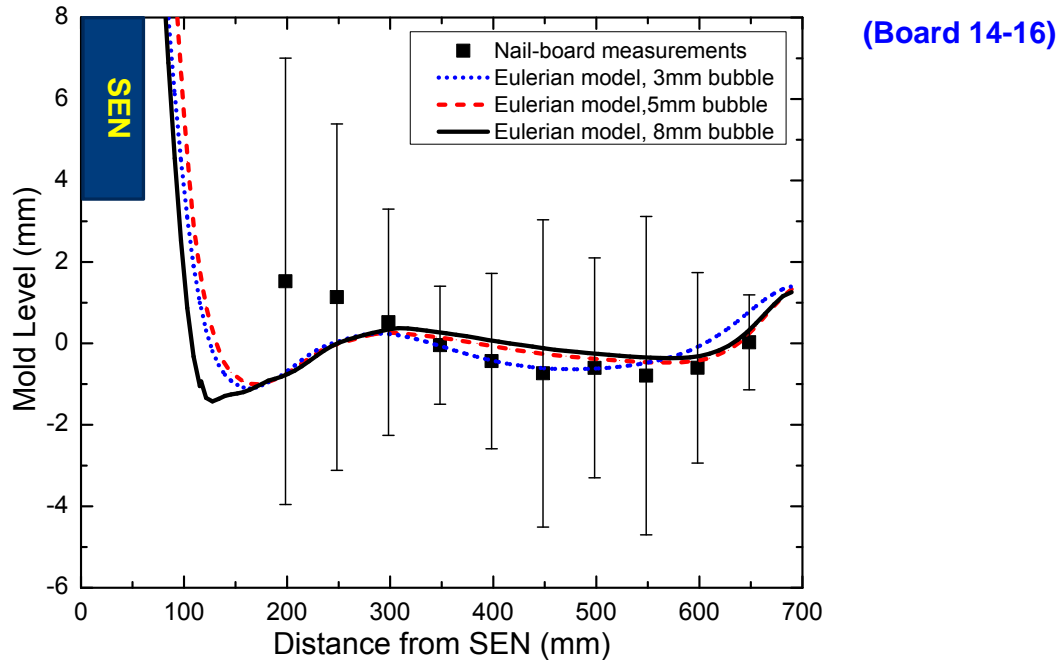


## Comparison with Nail Board 14~16 Measurements – Horizontal Velocity



(Board 14-16)

## Comparison with Nail Board 14~16 Measurements – Mold Level



## Conclusion – from BOARD 14~16

- **Excellent match is found between the measured and predicted surface steel velocities, as well as for the mold level, for all three bubble diameters used (3, 5 and 8 mm);**
- **Detailed comparison between simulations and measurements indicates that utilization of multi-size bubble groups instead of a single mean bubble size would increase the accuracy.**

# Summary and Future Work

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- **Methodology of modeling and simulating argon-steel multi-phase nozzle/mold flows has been established**
- **Models have been validated with many measurements and reasonable accuracy has been achieved for flow patterns, including complex flows with partial surface reversals due to gas bubbles rising**
- **Future work includes:**
  - **Develop and apply criteria for defects formation**
  - **Parametric studies to classify flows**
  - **Use methodology and models in this work to find operation windows to avoid defects**